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VOL III. Shot 10

PHOTOGRAMMETRY OF THE PARTICLE TRAJECTORIES ON DIPOLE WEST TRAJECTORIES ON DIPO

SHOTS 8, 9, 10, AND 11

Volume IV—Shot 11

University of Victoria

British Columbia

Canada V8W 2Y2

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14 MONITORING AGENCY NAME & ADDRESS(14-044) lerent from Controlling Office) 15. SECURITY CLASS (of this report) UNCLASSIFIED Defense Nuclear Agency 15a. DECLASSIFICATION/DOWNGRADING SCHEDULE Washington, D.C. 20305 16. DISTRIBUTION STATEMENT (of this Report) Approved for public release; distribution unlimited. 17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report) 18. SUPPLEMENTARY NOTES This work sponsored by the Defense Nuclear Agency under RDT&E RMSS Code B342077464 N99QAXAA11114 H2590D. 19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Photogrammetric Analysis Simultaneous Detonations Particle Trajectories Multiburst Detonations Air Particle Tracers This volume 20. ABSTRACT (Continue on reverse side if necessary and identify by block number) Volume 4 of this report describes the photogrammetry and analysis of the particle trajectories in blast waves produced by the simultaneous detonation of two spherical 1080-1b (490-kg) Pentolite charges (DIPOLE WEST_Shot 11). One of the charges was positioned at a height of 25 feet (7.6 m) above smooth ground, and the second charge 50 feet (15.2 m) above the first. Photogrammetrical measurements were made of the trajectories of air particle flow

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tracers (smoke puffs) which had been placed in a vertical grid at heights

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20, ABSTRACT (Continued)

Franging from 3 feet (0.92 m) to 58 feet (17.7 m) above the ground and at radial distances ranging from 25 feet (7.6 m) to 140 feet (42.7 m) from the vertical axis through the charges. From the measured particle trajectories, calculations were made of the particle velocities, densities, hydrostatic overpressures, dynamic pressures, and total pressures throughout the blast wave, at times ranging from 3 ms to 110 ms after detonation of the charges. The shock front times-of-arrival were also determined from the photogrammetrical measurements for the primary shock from each of the two charges, for the Mach stems produced above and below the interaction plane midway between the two charges, and for the Mach stem produced at the ground surface. From the shock front times-of-arrival, calculations were made of the shock velocities and, in turn, the peak particle velocities, air densities and hydrostatic overpressure immediately behind each shock. Calculations were also made of the variation with time of the particle velocity, density, hydrostatic overpressure, dynamic pressure, and total pressure at several fixed points. Results, presented both graphically and in tables, are compared to results previously calculated for the same experiment using shock front photogrammetry (Dewey, et al, 1975) and to measurements of side-on and total pressure obtained by electronic transducers (Keefer and Reisler, 1975). The analytical procedures used were similar to those described in Volume 1 (Dewey, et al, 1977).

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Owing to the quantity of material to be presented, this report is divided into several volumes. Volume 1 introduced the series, described the analytical procedures in detail, and presented and discussed the results for Shot 10. Volumes 2 and 3 presented and discussed the results for Shots 9 and 8. This volume presents and discusses the results for Shot 11. A subsequent volume will compare the results of the four experiments. The method of analysis is common to all four experiments and is described in detail in Volume 1 only.

So that the results from the four experiments may be easily compared, they have been scaled to remove the effects of varying atmospheric conditions. (Results are scaled to a 1 kg charge weight and a standard atmosphere of dry air at 15°C at sea level.) For the most part, only scaled results are presented. Exceptions include some derived pressure-time histories, which are compared to actual gauge measurements made in the experiment.

Results are presented in SI units, even though the experiments were originally laid out in British units. Only distance and time measurements are affected, however, as velocity, density, and pressure results are presented as dimensionless ratios. A distance units conversion scale is included to convert between SI units (meters scaled to a 1 kg charge) and British units (feet scaled to a 1 lb charge),

plus a time scale factor and scale factors to convert pressure ratios to both British and SI pressure units. Scale factors which may be used to compute the distance and time values actually observed under the ambient conditions of each shot are also provided. Dimensional pressure units are used for the results presented at gauge locations.

PREFACE

The authors gratefully acknowledge the opportunity offered by the Defence Research Establishment Suffield and the Defence Nuclear Agency to participate in the experiments described in this report. The analyses described here were carried out under contract with the Canadian General Electric Company, and with additional financial support from a research grant by the National Research Council (A 2952). The advice and assistance of Mr. A.P. Lambert, C.G.E. Project Officer at DRES, Dr. L. Kennedy, of the General Electric Company, and Mr. J. Keefer, of the Ballistic Research Laboratory, is also gratefully acknowledged.

Unit conversion and scaling factors

FEET (SCALING TO 1 LB CHARGE)

2	5.0
1 18	4.5
-	0.1
10	
σ	3.5
Ø	3.0
7	2.5
9	
S	2.0
	1.5
က	1.0
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0	3E 0.

METERS (SCALING TO 1 KG CHARGE)

For time scaled to a 1000 lb charge, multiply time scaled to a 1 kg charge by 8.683. For feet scaled to a 1000 lb charge, multiply the top scale by 10.

For pressure in kPa, multiply a pressure ratio (in atmospheres) by 101.325. For pressure in psi, multiply the pressure ratio by 14.696. To convert kPa to psi, divide by 6.895.

values in this report by 8.0730. To obtain the observed distance values in feet, multiply the reported scaled values by 26.485. To obtain observed time values, multiply scaled time values by 8.5933. For observed pressures in kPa, multiply by 94.34; for observed To obtain distance values actually observed for Shot 11, in meters, multiply scaled pressures in psi, multiply by 13.683.

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footnote

To assist in the comparison between volumes, similar tables have been numbered identically. For this reason table number 6 is not used in this volume.

CHAPTER 1, SHOT 11 ANALYSIS

1.1 Introduction

This is the fourth volume in a series which presents the particle trajectory analysis results from four experiments (Dipole West Shots 8, 9, 10 and 11) carried out to obtain information on the interaction of spherical blast waves with real and ideal reflecting surfaces. A general description of the project can be found in Volume 1. The results presented in this volume are for Shot 11.

In each experiment, photogrammetrical studies were made of the shock fronts (refractive image analysis, RIA), and of the motions of smoke puff particle tracers (particle trajectory analysis, PTA). The refractive image analysis results were reported by Dewey et al. (1975) and results of the particle trajectory analysis are presented in this report. The method of particle trajectory analysis, common to all four shots is described in detail in Volume 1 only.

1.2 Description of Shot 11

Dipole West Shot 11 was fired on November 8th, 1973 by
the Ballistics Research Laboratories at the Defence Research
Establishment Suffield, in Alberta, Canada. Two 1080 lb (490 kg)
spheres of Pentolite were detonated simultaneously, to within
5 microseconds, at nominal charge heights of 25 and 75ft
(7.6 and 22.9m) over a relatively rough ground surface.

Particle trajectory data were gathered by photographing the movement of smoke puffs formed in a vertical plane running out from ground zero at 6.7° south of west. A WF5 camera operating at about 3500 frames per second was positioned 30ft (9.2m) above ground level at a position 600ft (183m) due south of ground zero (GZ), the point on the ground vertically beneath the charges.

Table 1 gives the field survey data for the event, and Figure 1 shows a plan view of the layout. The dashed line represents the approximate line of sight of the WF5 camera. Figure 2 shows the field of view of this camera.

The smoke puff grid was made up of 20 columns of 12 puffs each, hung vertically on strings. The vertical spacing of puffs was 5ft, beginning 3ft above ground level and ending at a height of 58ft. The horizontal spacing of the columns of puffs was 10, 7 or 5ft, depending on the distance from ground zero, beginning at about 25ft and ending at about 140ft from GZ. Of the possible 240 smoke puffs, 232 detonated successfully. A good film record was obtained, except that several of the white smoke puffs in the bottom row could not be very easily distinguished against the background of snow-covered, ploughed earth.

This report describes the analysis of the smoke puff data collected for Shot 11, and presents and discusses some of the results of that analysis.

1.3 Camera calibration and data reduction

The calculated camera position coordinates and orientation angles for Shot 11 are presented in Table 2, together with the positions of photomarkers transformed from one frame of the film just before detonation to an object plane defined as passing through ground zero and being normal to the camera orientation axis. The differences ("shifts") between the object plane positions of the transformed calibration points and their positions computed from the field survey data are given in Table 2. The object plane positions of the calibration points computed in these two ways are also shown in Figure 3.

The camera calibration procedure, described in detail in Volume 1, ensured that selected photomarker images (Pl to P5) transformed to the object plane in a way which matched them exactly to the positions computed using the survey data. These reference photomarkers for Shot 11 are indicated in Figure 3 using large circles: namely, Pl = W1, P2 = W3, and P3 = 300Wl. The separation distance between P4 = P3 = 300Wl and P5 = 300W2 was also used as a calibration parameter. The probable reason for the shifts seen for photomarkers VP1A and VP1B was discussed in Volume 1.

The image positions of two reference photomarkers

(VP3B and 300W2) and all smoke puffs were measured frame-byframe over a time interval corresponding to the approximate

duration of the positive phase of the blast waves (film frames 10 to 375), and were transformed to distances in the object plane by matching the reference marker positions to their positions transformed from the calibration frame. These data were again transformed from the object plane to the smoke puff plane which was assumed to pass through "corrected" ground zero; to be vertical, and to run 6.7° south of west from GZ.

The x-y coordinate system in the smoke puff plane was the same for Shot 11 as for the other shots, except that the corrected value for ground zero was displaced 0.7ft from the surveyed ground zero, in a direction approximately 34° south of west. The corrected ground zero was defined to have the same elevation as the surveyed ground zero, but was located directly under the midway point between the two charge centers. As for the other shots, all data in the output plane are plotted with the x coordinate reflected, i.e. with positive values of x to the right hand side, as if the smoke grid had run to the right of the charges rather than to the left as seen in the film images.

A time was assigned to each film frame using the 1 ms timing marks placed on the film during its exposure. The film timing method was described in Volume 1, and the complete set of film timing data used for Shot 11 is provided in Table 3.

Figure 4 shows the positions of the detonated smoke puffs at a time prior to the detonation of the two charges. These

positions are in the plane of the charges and the smoke puff grid, as described above. The smoke puff plane was not exactly parallel to the camera image and object planes (Figures 2 and 3), and various geometrical corrections were applied to make the transformation between them. The puffs enclosed in parenthesis were not visible in the earlier film frames because they were concealed by the photomarkers, but were seen later. The puffs which are underlined in the figure were seen in their initial positions only - no trajectory data could be obtained. Charge positions in the figures are plotted as if they were positioned exactly above the corrected ground zero origin. The data shown in Figure 4 have not been scaled.

1.4 Data scaling and trajectory fitting

The position-time histories of individual smoke puffs were extracted from the frame-by-frame positions of the smoke puff grid, and then scaled to standard atmospheric conditions and charge weight. A change to SI units was made at this point in the analysis. The resulting trajectories were edited, and then smoothed by fitting polynomial functions.

Particle trajectory data were scaled by dividing all distances by Sachs scaling factor $S = \sqrt[3]{(WP_O)/(W_OP)}$ and multiplying all times by the factor $C/(C_OS)$, where C is the ambient sound speed computed for Shot 11. Data used to compute

C and S, and define the scaled event, are listed below with the computed values of C and S.

Ambient temperature,	T = -19.11 °C	(-2.4 °F)
Ambient pressure,	P = 94.34 kPa	(13.683 PSI)
Relative humidity,	RH = 60.0%	
Computed vapour pressure,	VP = 0.08 kPa	(0.6 mm Hg)
Ambient sound speed	C = 319.689 m/s	(1049 ft/s)
Charge weight,	W = 489.9 kg	(1080 lbs)
Sachs scaling factor	S = 8.0730	
Standard charge weight,	$W_0 = 1.0 \text{ kg}$	(2.2 lbs)
Standard pressure,	$P_{O} = 101.325 \text{ kPa}$	(14.7 PSI)
Standard temperature,	$T_O = 15 ^{\circ}C$	(59 °F)
Standard sound speed, (dry air)	$C_{O} = 340.292 \text{ m/s}$	(1116 ft/s)

The results presented in this report therefore apply to a scaled event which is the detonation of two 1 kg charges in a standard atmosphere. The scaled heights of burst for Shot 11 were 0.905 m and 2.797 m, and the scaled charge separation divided by two, was 0.946 m.

Figure 5 shows the scaled particle trajectory data for Shot 11 in the smoke puff plane with positions measured horizontally and vertically from corrected ground zero. Approximately 27,247 puff positions are represented. As represented, the raw trajectory data have not been smoothed.

The raw particle trajectory data were edited to remove obvious data processing errors, such as a single point widely

displaced from its trajectory for one or two frames. The trajectory of each puff in turn was then smoothed by least squares fitting simple polynomial expressions separately to both the x and y coordinate data, these being discrete functions of frame time. The adequacy of each fit was determined by examining on the same graphical output, plots of both the raw trajectory data and the fitted curve. For Shot 11 this meant examining and adjusting 464 such plots, at least two or three times each.

For a given puff, the first step in fitting the raw trajectory data was to set the time of arrival of the shock front first hitting the puff. The data at subsequent times were fitted with polynomial functions, as described in Volume 1, paragraph 2.5. The first derivatives of the fitted functions were also calculated at a series of times for use in later calculations of particle velocity.

1.5 Regionalization and shock strength calculations

Five regions were defined in the smoke puff plane on the basis of the shock front which first struck the puffs in a particular region. These are shown in Figure 6. The regions were bounded by the triple point trajectories measured using refractive image analysis (Dewey et al., 1975). Regions 1 and 2 are those in which the smoke puffs were first hit by a spherical primary shock front, and regions 3, 4, and 5 are those in which the puffs were first hit by a

Mach stem.

In each of the five regions, the shock trajectory data obtained from the first movement of the smoke puffs were fitted to a function of the form

 $r(t) = A + Bt + C \log (1 + t) + D\sqrt{\log (1 + t)} \;,$ where r is the shock radius, t is the time after detonation, and A, B, C and D are the fitted coefficients. The shock velocities were calculated by differentiating this function. The peak particle velocity, $V_{_{\rm S}}$, peak density, $D_{_{\rm S}}$, and peak hydrostatic overpressure, $P_{_{\rm S}}$, as functions of shock radius in each of the five regions, were calculated from the shock velocity using extensions of the Rankine-Hugoniot equation. Details of the shock radius calculations etc. are described in Volume 1, paragraph 2.6.

1.6 Particle velocity calculations

Particle velocities were computed using the methods described in Volume 1, paragraph 2.7.

1.7 Density and hydrostatic overpressure calculations

Densities and hydrostatic overpressures in the smoke puff plane were calculated by the method described in Volume 1, paragraph 2.8. Results in both cases represent average values over cells defined by four adjacent smoke puffs.

1.8 Surface representation

Surfaces were fitted to the times of shock front arrival and to the fields of particle velocity, density and hydrostatic overpressure at a sequence of times. All data were interpolated onto a common regular Euleurian grid. Fields of dynamic pressure were computed from surface-interpolated particle velocity and density results. Contour plots were generated for all surfaces at selected times, and time histories computed at several fixed locations. The methods used were identical to those described for Shot 10 in Volume 1, Chapter 3.

1.9 Pressure and total pressure time-histories

To permit a direct comparison between results obtained from the particle trajectory analysis and measurements made using side-on and face-on pressure transducers, the hydrostatic and total overpressure time-histories were calculated at those locations coincident with gauge positions within the smoke puff grid. Dynamic pressures and hydrostatic overpressures obtained from the particle trajectory analysis were used to compute the total pressures after application of a compressibility correction. This correction is a function of the local Mach number and its form depends on whether the Mach number was greater or less than unity. The time varying hydrostatic and total overpressure impulses, determined by integrating the pressure time histories, were also calculated

and compared with similar integrations of the electronic transducer data.

The methods used to calculate the total pressures and the impulses are described in detail in the addendum to Volumes 1 and 2, which is incorporated in Volume 3. In cases where the leading edge of a time-history curve was rounded, impulse integrations were done using data interpolated linearly between the peak parameter value determined at the time of arrival, and a point on the time-history curve subsequent to the time of arrival. The second of these two points was chosen in a manner which ensured a minimum difference in slope between the interpolated and computed sections of the time-history data.

CHAPTER 2. SHOT 11 RESULTS

2.1 Times of shock front arrival

The measured initial puff positions, the times of first shock arrival, and the peak particle velocities obtained by differentiating the functions fitted to the particle trajectories are presented in Table 4. Puff position is given relative to corrected ground zero as origin, with horizontal and vertical axes. Puff position and the time of arrival of the first shock are given both as observed and scaled. Particle velocities listed are derivatives of the fitted puff trajectories at the times of shock arrival, and are expressed in Mach units. Expressed this way, the particle velocities are the same scaled as unscaled. Also listed are the initial radial puff positions (scaled) and region codes.

Shock front data determined from the first movement of the smoke puffs, i.e. calculated from the time-of-arrival data in Table 4, are listed in Tables 5.1 - 5.5. Each table corresponds to one of the 5 regions used. Listed are the observed and fitted unscaled shock trajectory data, the scaled fitted shock trajectory data, and the computed shock velocities and peak parameters associated with shock strength: peak hydrostatic overpressure in atmospheres and kilopascals, peak

particle velocities in Mach units, and peak density ratios. Given as ratios, these peak parameters are the same scaled as unscaled. Pressure given in kilopascals in the tables refers to the unscaled (observed) case only.

The shock front radius versus time data derived using particle trajectory analysis (PTA) are also shown in Figures 7.1-7.3 for the two primary fronts, the two Mach stems at the interaction plane, and the ground Mach stem, respectively. They are compared to corresponding data derived from refractive image analysis (RIA) reported by Dewey et al. (1975). The refractive image analysis results were obtained using photogrammetry against a striped canvas backdrop and they describe the shock as it travelled in a direction almost diametrically opposite to the direction of the smoke puff grid.

2.2 Shock strengths

Peak particle velocities calculated from shock front velocities are shown in Figures 8.1 - 8.3 for the primary fronts, interaction Mach stems, and the ground Mach stem. This method of determining peak particle velocities was labelled method 1, and the data plotted correspond to those listed in Tables 5.1 - 5.5. The results in the figures are compared with those previously obtained using refractive image analysis (RIA). In the case of the primary shock fronts, results are also compared to those of Brode (1957) for TNT.

In Volume 1 other methods of determining shock strengths in the various regions were described. It was demonstrated that method 1 was clearly the most accurate, and in the present volume shock strengths calculated using methods 2 and 3 are not reported. For this reason Figures 9, 10 and 11 and Table 6 do not appear in this volume.

2.3 Particle velocity fields

The calculated particle velocities in the plane of the smoke puffs are shown as vectors in Figures 12.1 through 12.8, for various times after the detonation. All times and positions are scaled to a 1 kg charge in a standard atmosphere. The particle velocity vectors represent the derivatives of the smoothed particle trajectories, and their magnitudes may be judged using the standard vector shown on each figure. All velocities are measured in Mach units, relative to the standard sound speed. Puffs not yet struck by a shock wave are represented by small circles (zero velocity).

Numerical data corresponding to Figures 12.1 - 12.8 are listed in Tables 7.1 through 7.11, along with scaled radial positions of the puffs, and region codes as defined in Figure 6. Conversion factors are given at the foot of each table, which may be used to convert the scaled data in the tables and figures back to their original unscaled values.

2.4 Density and hydrostatic overpressure fields

Calculated average relative densities throughout the smoke puff plane are depicted graphically in Figures 13.1 - 13.4, for various times after the detonation. All time values are scaled. Cell positions are scaled and are given relative to the corrected ground zero as origin with horizontal and vertical axes. The calculated densities may be judged using the density shading scale shown on each figure. Density is given as a ratio, relative to ambient density. Cells not yet struck by a shock wave and cells in which the density has dropped to a value less than ambient density are shown blank.

Corresponding numerical data are listed in Tables 8.1 - 8.8 along with radial cell positions computed according to the regions defined previously. Numerical data describing the fields of hydrostatic overpressure are similarly listed in Tables 9.1 - 9.8. The pressure results for a given cell were obtained by multiplying the density results for that cell by a factor determined by the strength of the shock which first traversed the cell and which then remained constant, i.e. by assuming isentropic flow after the first shock.

2.5 Times-of-arrival surface

Figure 14 shows a perspective view of the surface fitted to the original smoke puff positions and the observed times of first shock front arrival, i.e., to the data listed in Table 4.

The grid mesh size is 0.1 by 0.1 meters (scaled), about 2.5 feet square (unscaled), or about half that of the original smoke puff grid. The charge positions are indicated on the vertical distance axis.

The times-of-arrival surface is smooth enough to permit contouring, the contours in this case (isochrones) representing shock front shapes at different times, as shown in Figure 15, but the surface is not smooth enough to permit the calculation of gradient vectors which could be used to compute shock velocity vectors and shock strengths over the new grid.

Two attempts were made to obtain contours of shock strength. In the first, the times-of-arrival surface was smoothed by least-squares fitting low-order, one-dimensional polynomial functions to the time-of-arrival data along each grid row and column separately, and computing the derivatives of the fitted functions to obtain the associated components of the surface gradient vectors. Shock velocity vectors were obtained from the time-of-arrival gradients, and from these peak particle velocities were computed. The peak particle velocity (shock strength) surface is shown in Figure 16. The contours of this surface (not shown) did not exhibit any discontinuities across the boundaries of the shock front regions, as they would if surfaces had been fitted to the time of arrival in each region separately.

The results of a second method used to compute shock strength contours are shown in Figure 17. These were obtained

by interpolating shock radius at each value of peak particle velocity shown, for each shock front region in turn, using the peak particle velocity versus radius curves shown in Figures 8.1 - 8.3. Arcs of circles with these radii, centered on the appropriate points along the vertical charge axis, were then drawn in the regions to represent shock strength contours. These peak value contours are discontinuous across triple point locii and other region boundaries. As a result, some horizontal lines are crossed twice by the same contour or, in other words, indentical shock strengths can be found at two locations the same vertical distance from a reflecting surface, but at different radial distances from the vertical charge axis.

2.6 Field surface contours

Contours of equal particle velocity, density, hydrostatic overpressure, and dynamic pressure in the blast waves were determined for a series of times, using surfaces fitted to the various measured data fields at those times. Sample results are shown in Figures 18 through 21 at scaled times of 2.5, 4.0 and 9.0 ms. The shock fronts shown in these figures are obtained from the time-of-arrival surface (as were those in Figure 15). Field contours such as those shown can be drawn for any scaled time between 0.5 ms and 12.7 ms.

It should be re-stated that all of these results were obtained from the photography of the smoke puffs only and do

not rely on the results obtained using the refractive image analysis (Dewey et al., 1975).

2.7 Time histories

By mapping the physical properties of the blast waves at short time intervals its was possible to determine the time histories of these properites at any selected fixed position within the smoke puff grid. This was done at 15 fixed locations, three in the primary region of the lower charge and four in each of the three Mach stem regions, as shown in Figure 22. At each distance from the vertical axis through the charges in the Mach stem regions, each of the time history stations is approximately the same distance from either the interaction plane or the ground plane. (Particle velocity time histories could be interpolated closest to the grid edges because these were measured at puff locations, whereas the density and pressure data were measured at cell centers).

Time histories of particle velocity, density, hydrostatic overpressure and dynamic pressure at these locations are given in Figures 23 to 26. Time-histories of these physical properties of the blast wave can be provided at any other location within the smoke puff grid, on request.

The vertical line which forms the leading edge of a time-history plot represents the interpolated time-of-arrival of the first shock at the given location, and the height of this line represents the peak parameter value derived from the shock velocity analysis.

The dynamic pressures plotted in Figure 26 are maximum values, computed using both the x and y component of particle velocity. Similar plots were made of the horizontal components of dynamic pressure, but the differences were not significant since the y components of particle velocity at these locations were small. Other locations could have been chosen at which the y components would not have been insignificant.

Time histories for hydrostatic overpressure and total pressure are also plotted in Figures 27.1 to 27.4 for stations at the nominal positions of field-mounted pressure gauges on the "60 foot gun barrel". The gauges on this gun barrel were mounted at nominal elevations of 10, 20, 30, 40, 47, 50 and 53 feet. The time histories at these locations are compared to the gauge measurements (Keefer and Reisler, 1975). The total pressures were calculated in the manner described in the addendum to Volumes 1 and 2 which is incorporated in Volume 3. The variation with time of the integrated pressure (pressure impulse) is also shown in these figures, compared with similar integrals of the gauge data (Keefer and Reisler, 1975).

CHAPTER 3, DISCUSSION

3.1 Particle trajectory analysis, Shot 11

The methods used to analyze the smoke puff trajectories on Shot 11 were identical to those used for Shots 8, 9 and 10 and described in detail in the first three volumes of this report. The results for Shot 10 clearly indicated the superiority of one of several methods of analyzing shock strength, and only the results of this method have been reported for Shots 8, 9 and 11.

3.2 Primary shock strength of upper charge

The refractive image analysis of the shock fronts described by Dewey et al (1975) did not provide any information about the primary spherical shocks from the upper charges, and it was assumed that these charges had detonated satisfactorily. This assumption was validated for Shots 10 and 9 by the analysis of the particle trajectory time-of-arrival measurements. In Figure 7.1 the primary shock radii are plotted versus time for the upper and lower charges of Shot 11, together with the results obtained for the lower charge by the refractive image analysis. All three curves appear to be identical. The limited range of the data obtained for the primary shock from the upper charge did not permit an accurate calculation of the variation of the shock strength with distance. However, the results for the lower charge, in Figure 8.1,

show a similar shock strength variation with distance to that obtained from the refractive image analysis and one which is very similar to Brode's (1957) calculation for TNT.

3.3 Comparison of Mach shock strength over different surfaces

The refractive image analysis of Shot 11 (Dewey et al., 1975) showed what appeared to be a significant difference between the strengths of the Mach shocks over the rough ground and beneath the interaction plane between the two charges. The results of the particle trajectory analysis given in Figures 8.2 and 8.3 show a similar but smaller difference. The RIA measurements were made as close as possible to the reflecting surfaces, 0.5 m above the ground plane and 0.2 m below the interaction plane, whereas in the PTA case the results represent an average of measurements made at puff positions at heights ranging between 1.0 and 7.0 m. Obtaining particle trajectory data near the ground surface was a particular problem on Shot 11 since 13 of the 20 smoke puffs in the bottom row either failed to detonate or could not be distinguished in the film images (they were white smoke puffs placed against a background of the snow-covered, ploughed earth). The results in Figures 8.2 and 8.3 therefore indicate that the difference in shock strength over the ground compared with that at the interaction plane may be dependent on the height above the ground - not an unexpected result.

In addition, determination of Mach shock strength from measurements of the times of shock arrival at smoke puffs at various distances from the reflecting surfaces is difficult because an assumption must be made about the exact shape of the Mach shock front, in order to correctly assign shock radius values at smoke puff positions. At or near a reflecting surface the problem of shock shape is not so important as it is assumed that the shock is perpendicular to the surface. Details of this aspect in the PTA case and the manner in which the problem was dealt with for Shots 8 through ll are described in Volume 1 of this report.

The possible magnitude of the difference between Mach stem shock strength over the rough ground and at the interaction plane is best illustrated in Figure 17, which shows peak particle velocity isotachs derived from the RIA results presented in Figures 8.1 - 8.3. It can be seen, for example, that the Mach 0.5 isotach at the interaction plane is significantly farther from the vertical charge axis than it is at the ground plane. The apparent discontinuity in the Mach 0.5 isotach at the plane y = 0.8 cannot be maintained and there will be a transfer of energy between the two shock waves causing the upper Mach stem to slow down and the lower to strengthen.

3.4 Resolution of time histories

The time histories of density and pressure given in Figures 24 and 25 do not always show a sharp rise at the shock front. This is not a real effect but one inherent to the present method of particle trajectory analysis, which does not permit a high resolution of density in space or in time because the average density of the air within a rectangular cell defined by four smoke puffs cannot be calculated accurately until the shock has completely traversed the cell. The time of complete traversal may be as much as 5 ms, or 0.6 ms when scaled.

For the same reason the calculated time histories sometimes anticipate the time of shock front arrival (e.g. Fig. 24.4, position 4.0, 0.4) and do not resolve weaker shocks subsequent to the first, although these may be detected occasionally as a rounded bump in the normally exponentially decaying curve. Efforts are being made to improve the space and time resolution of density and pressure calculated from the particle trajectories.

The lack of resolution close to the shock front does not occur in the case of particle velocity, which can be measured with reasonable accuracy as soon as the shock has traversed the relatively small space represented by an individual smoke puff. This improved resolution is manifested also in the dynamic pressure histories which depend on particle velocity squared.

3.5 Comparisons with gauge results

The hydrostatic and total pressure time-histories were calculated at gauge positions within the smoke puff grid and the results from the particle trajectory analysis are compared with the corresponding transducer outputs in Figures 27.1 to 27.4.

The agreement between the results from the two measurement methods is good although, as previously discussed, the poor time resolution of the particle trajectory results does not permit indentification of multiple shocks. This is illustrated in Figure 27.1 for the 60.20 location. However, the agreement between the two impulse curves is excellent.

The good agreement between the total-pressure timehistories from the two measurement methods shown in Figures 27.3 and 27.4 further confirms the validity of the technique used to calculate the total pressure, as described in the addendum to Volumes 1 and 2 of this report.

In considering the above comparisons it must be remembered that determination of pressure time-histories was not an original objective of the particle trajectory analysis project, but the reasonable agreement with gauge measurements gives some indication of the reliability of the method. Also the gauge measurements were made on the opposite side of the charges to the smoke puff grid so that some differences might be expected due to slight non-symmetries of the blast waves.

Although the effect is more clearly seen from the gauge results rather than the particle trajectory analysis, it is interesting to note the difference in the blast wave signature at location 60.10, 10ft above the rough ground, compared with that at 60.40, 10ft below the interaction plane (Fig. 27.1). Above the ground the two initial shocks indicate that the gauge was above the triple point and detected the primary and the reflected shock. Below the interaction plane the single shock indicates that the gauge was in the Mach stem below the triple point. This confirms the photogrammetrical measurements (Dewey, et al., 1975) which showed the triple point above the ground at this location to be at height of approximately 5.8 ft and to be approximately 9.6 ft below the interaction plane, i.e., almost coinciding with the gauge position.

3.6 Comparisons of time histories at different heights

Dewey et al., (1975) pointed out that although the shock strength below the interaction plane at a given distance from the vertical charge axis was greater than it was above the rough ground, the gauge results showed that the hydrostatic pressure impulse was greater along the ground. In other words, along the ground the peak pressure value at the shock was less, but the decay of the pressure was slower and the duration longer.

The results from the particle trajectory analysis have been studied to see if these conclusions are confirmed. Unfortunately, gaps in the smoke puff grid close to the roughened ground, seen in figure 22, meant that fewer results were obtained in this region than had been hoped for. Nevertheless, a number of comparisons between time histories below the interaction plane and above the ground are possible. The appropriate time histories to be compared are the particle velocity profiles in figure 23.2 at (2.5, 1.6) and (2.5, 0.2), in figure 23.3 at (3.0, 1.6) and (3.0, 0.2), and in figure 23.4 at (4.0, 1.6) and (4.0, 0.3); the density profiles in figure 24.3 at (3.0, 1.6) and (3.0, 0.4) and in figure 24.4 at (4.0, 1.6) and (4.0, 0.4); the hydrostatic pressure profiles in figure 25.3 at (3.0, 1.6) and (3.0, 0.4), and in figure 25.4 at (4.0, 1.6) and (4.0, 0.4), and the dynamic pressure profiles in figure 26.3 at (3.0, 1.6) and (3.0, 0.4) and in figure 26.4 at (4.0, 1.6) and (4.0, 0.4). Although the differences are not great when the accuracy of these profiles is considered, it appears to be significant that in every one of the nine comparisons, close to the rough ground the shock front arrives later and the peak value is less than below the interaction plane, but the compared history profiles always cross so that in the latter part of the wave both the hydrostatic and the dynamic pressure are greater above the ground than at the interaction plane.

References

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 Photogrammetry of the Shock Front Trajectories on
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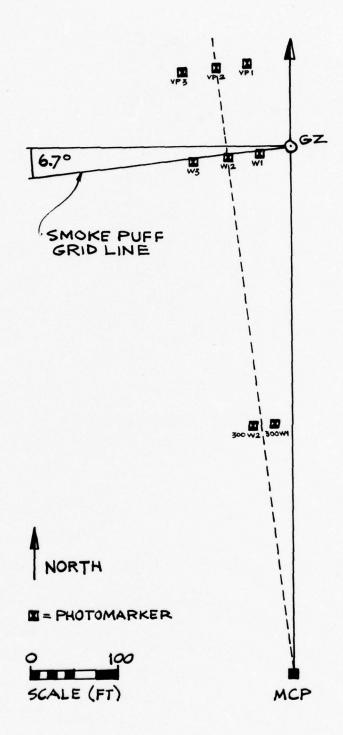


Fig. 1 Plan view of test site, Dipole West/ll

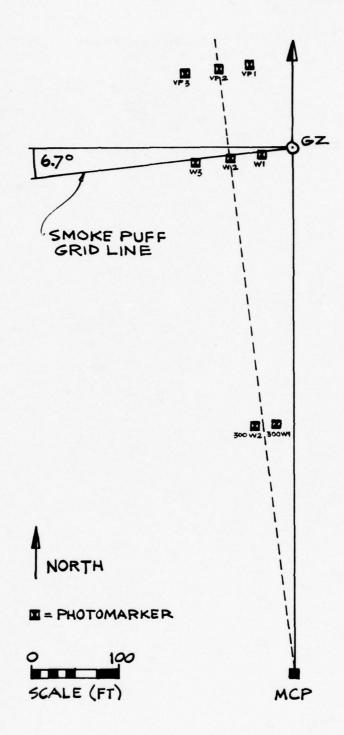
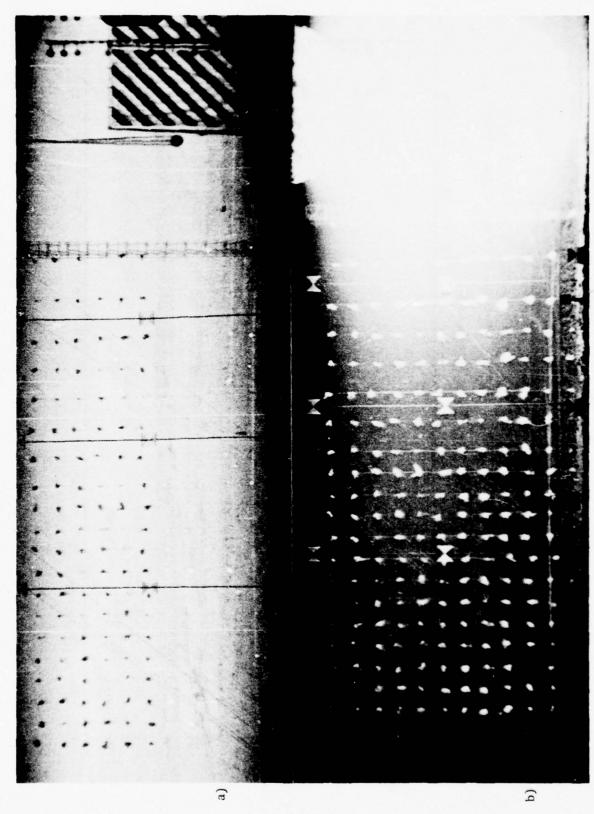
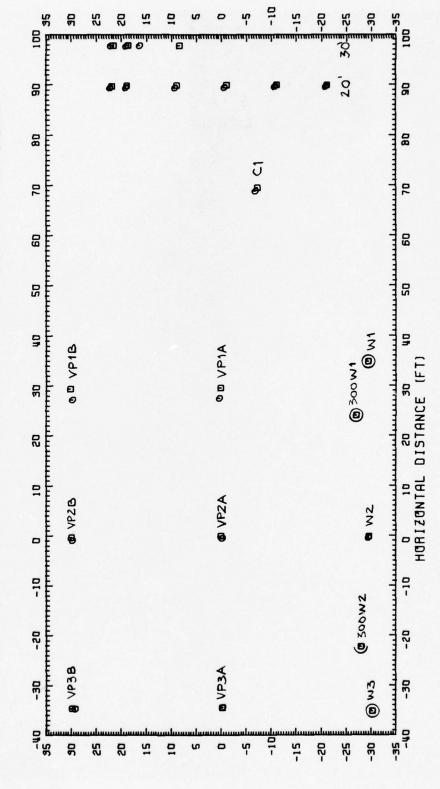


Fig. 1 Plan view of test site, Dipole West/ll



Field of View of Camera—Smoke puff grid, DIPOLE WEST/11: a) before detonation, b) approximately 3 ms after detonation.

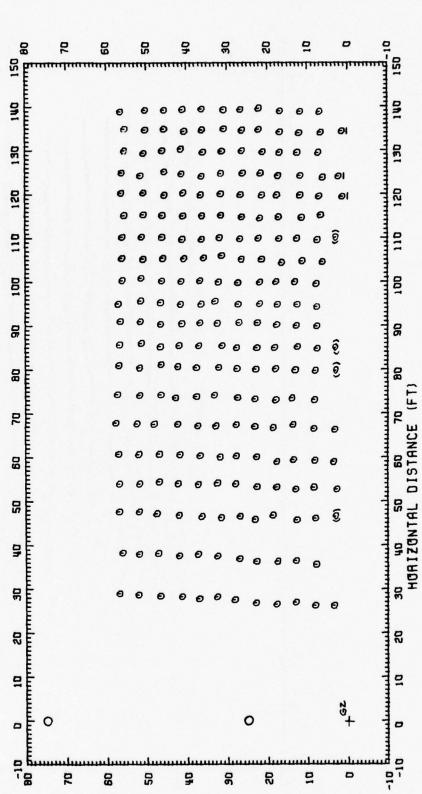
CHOTOMARKER POSITION IN OBJECT PLANE CALCULATED FROM SURVEY DATA 0 = PHOTOMARKER POSITION IN OBJECT PLANE TRANSFORMED FROM FILM IMAGE



VERTICAL DISTANCE

Fig. 3 CAMERA CALIBRATION, DIPOLE WEST/11

Fig. 4 SMOKE PUFF GRID, DIPOLE WEST/11



VERTICAL DISTANCE (FT)

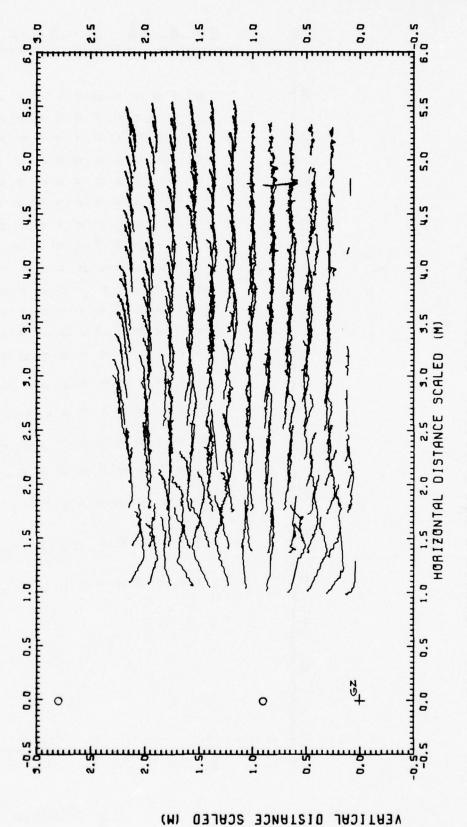
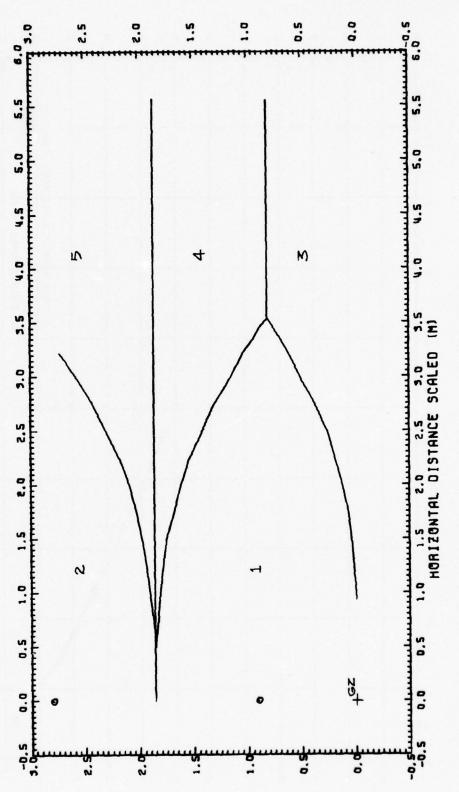


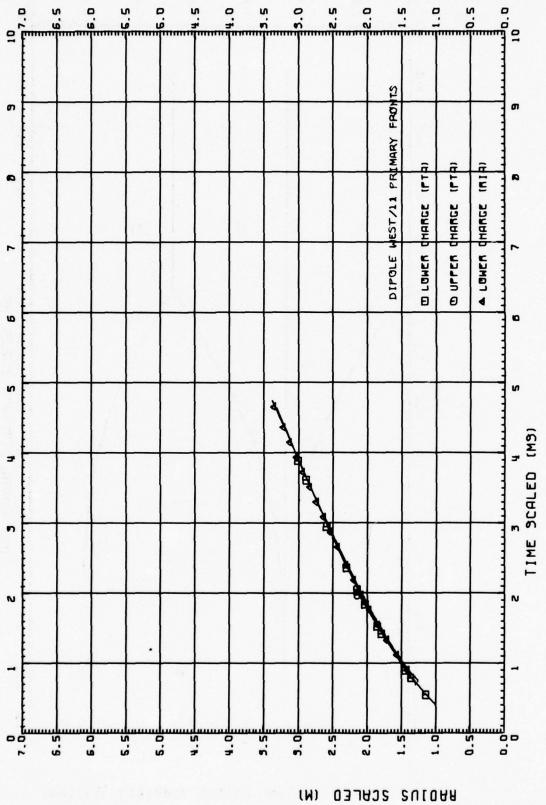
Fig. 5 PARTICLE TRAJECTORIES, DIPOLE WEST/11

VERTICAL DISTRNCE SCALED

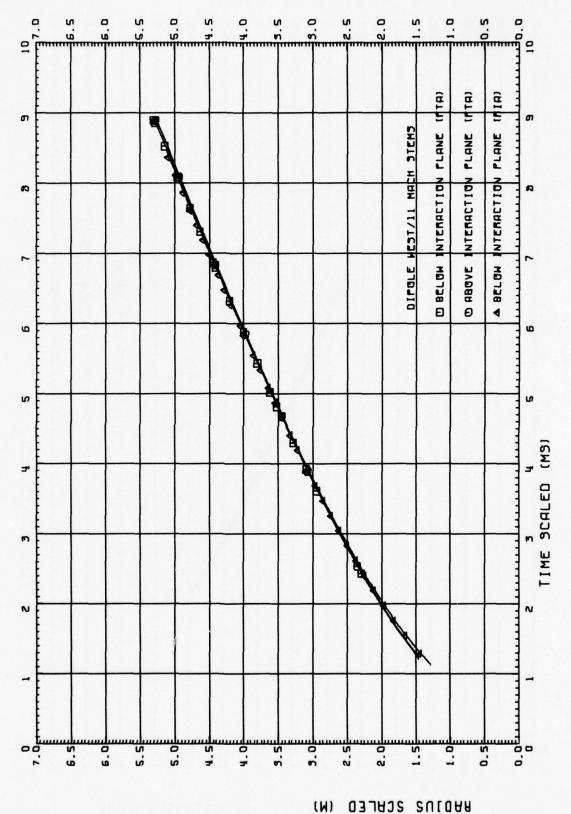


S REGIONS DEFINITION, DIPOLE WEST/11



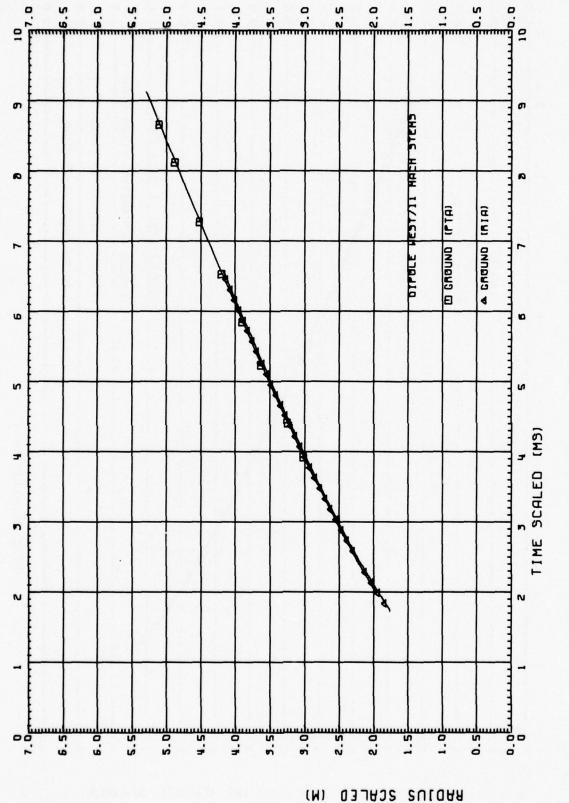


SHOCK TRAJECTORIES, DIPOLE WEST/11

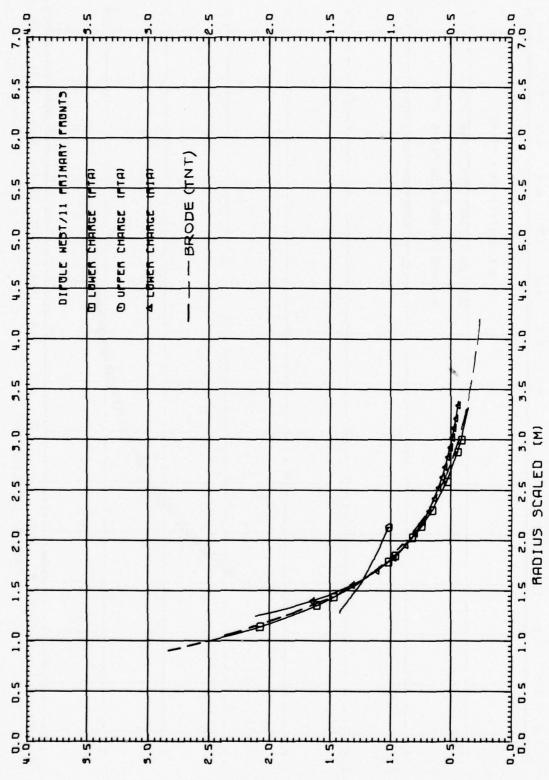


43





.3 SHOCK TRAJECTORIES, DIPOLE WEST/11



(RECH UNITS)

45

SHOCK STRENGTH, DIPOLE WEST/11

Fig. 8.1

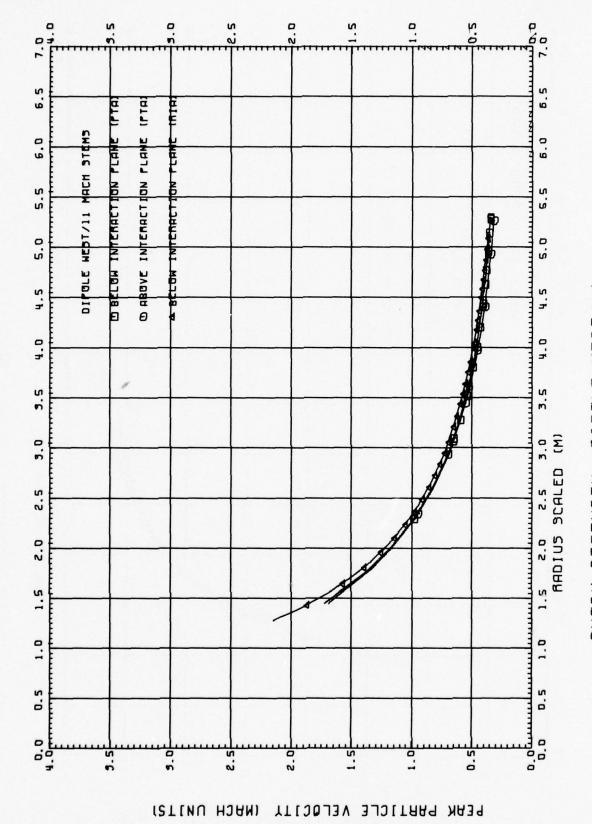
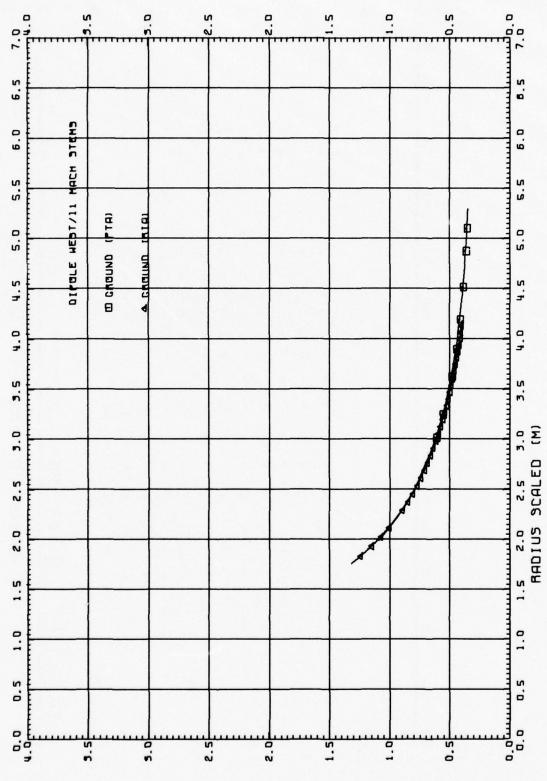
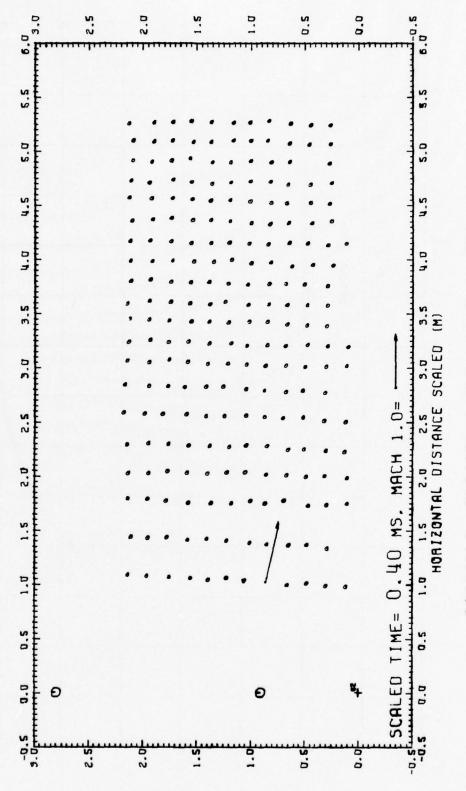


Fig. 8.2 SHOCK STRENGTH, DIPOLE WEST/11



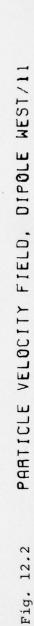


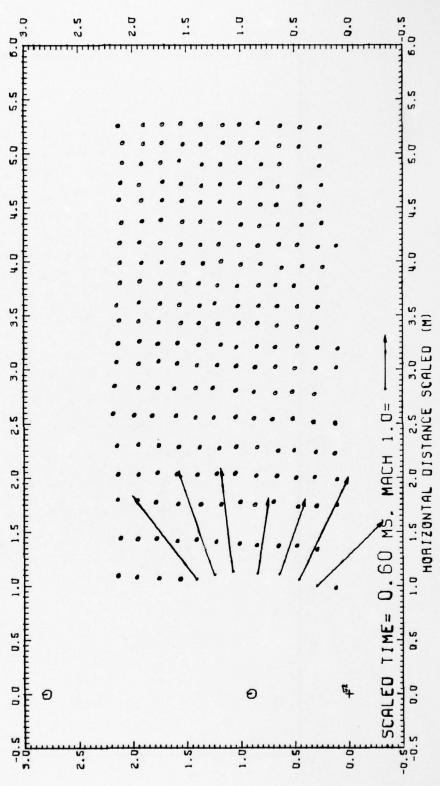
PERK PRRIJCLE VELOCITY (MRCH UNITS)



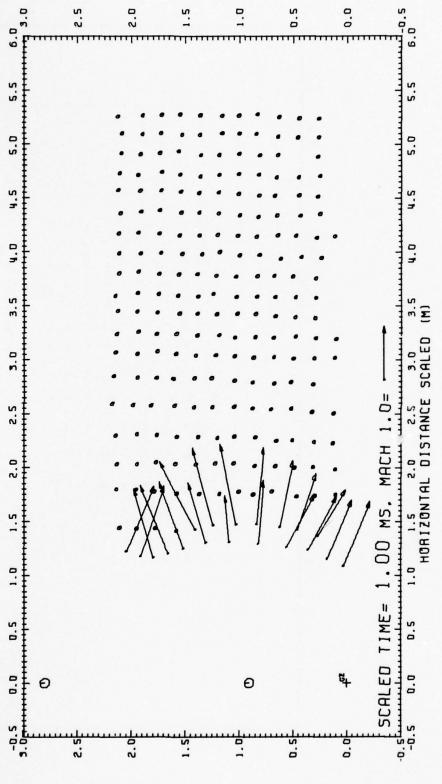
PARTICLE VELOCITY FIELD, DIPOLE WEST/11 Fig. 12.1

VERTICAL DISTANCE SCALED

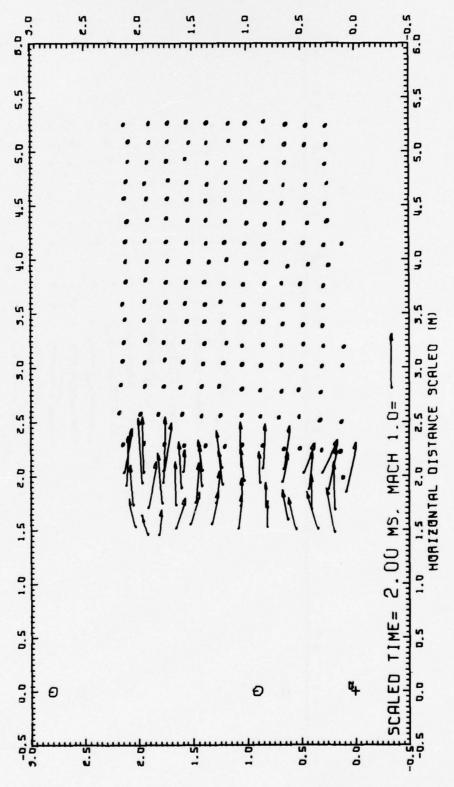




VERTICAL DISTANCE SCALED



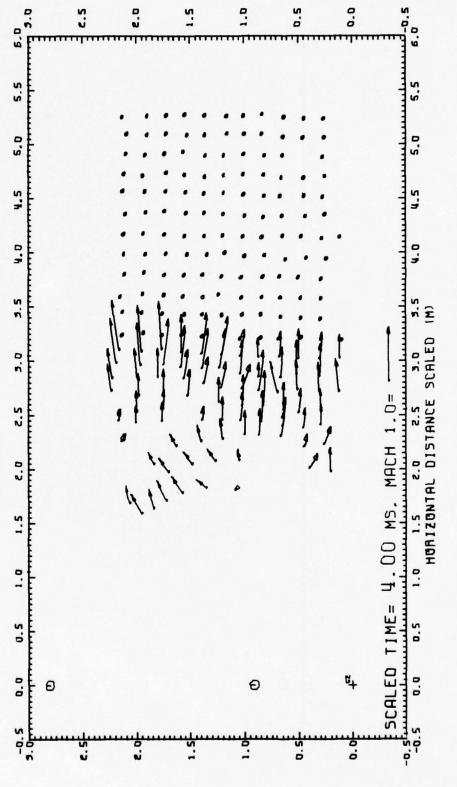
PARTICLE VELOCITY FIELD, DIPOLE WEST/11



The same

VERTICAL DISTANCE SCALED IM)

VERTICAL DISTANCE SCALED

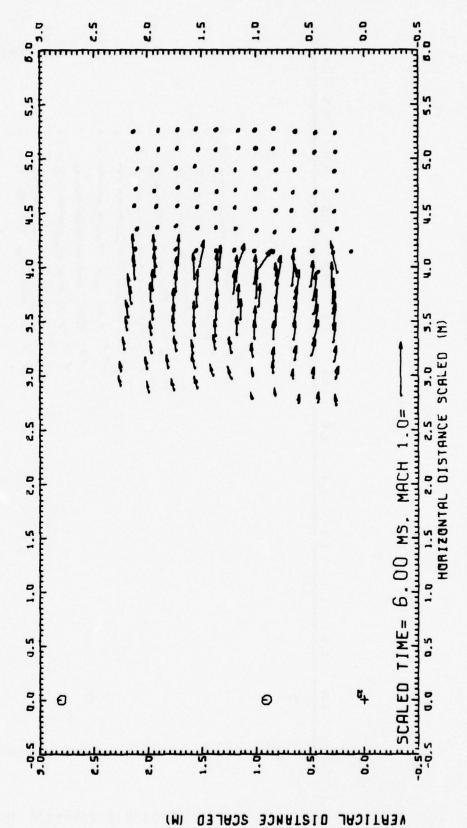


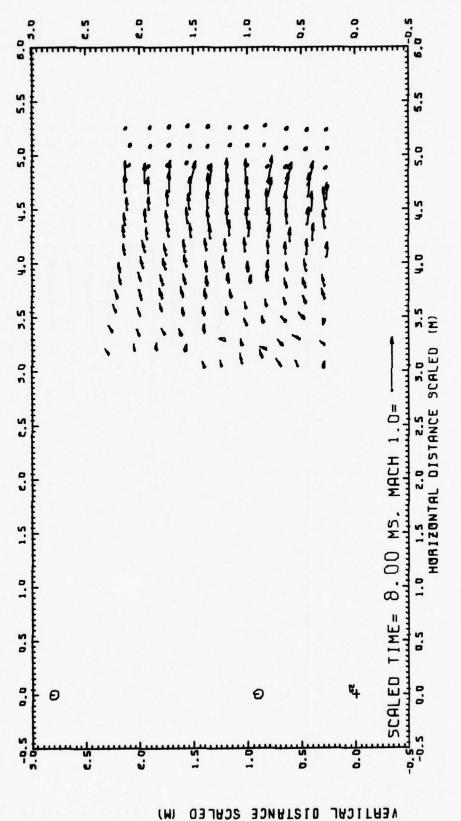
PARTICLE VELOCITY FIELD, DIPOLE WEST/11

Fig. 12.5

Fig. 12.6

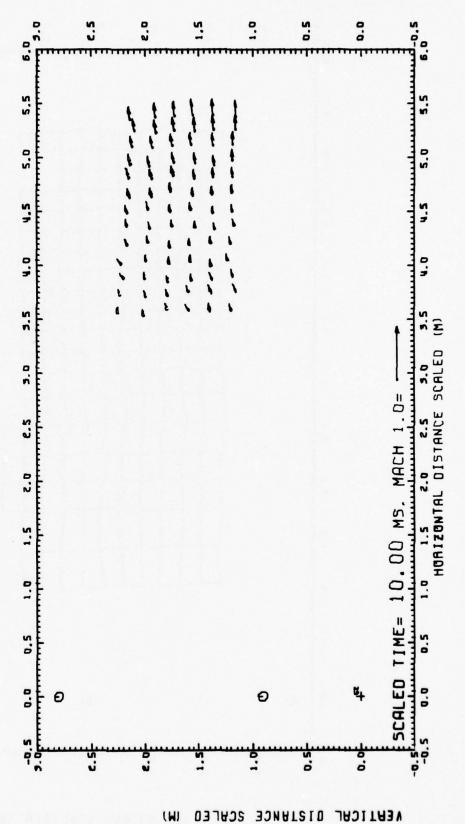


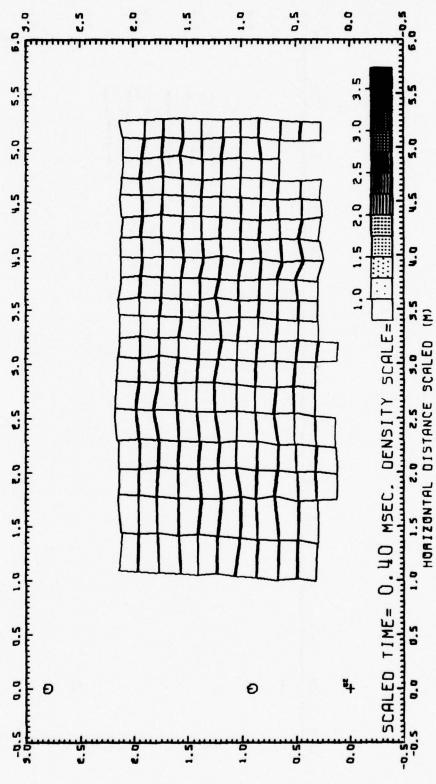




PARTICLE VELOCITY FIELD, DIPOLE WEST/11 Fig. 12.7



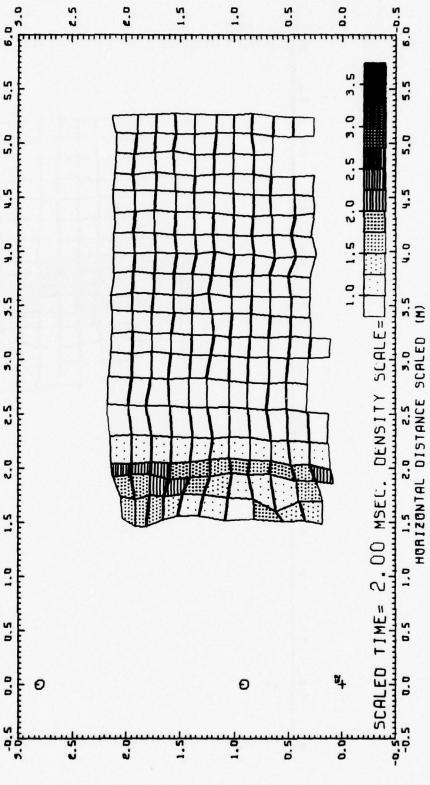




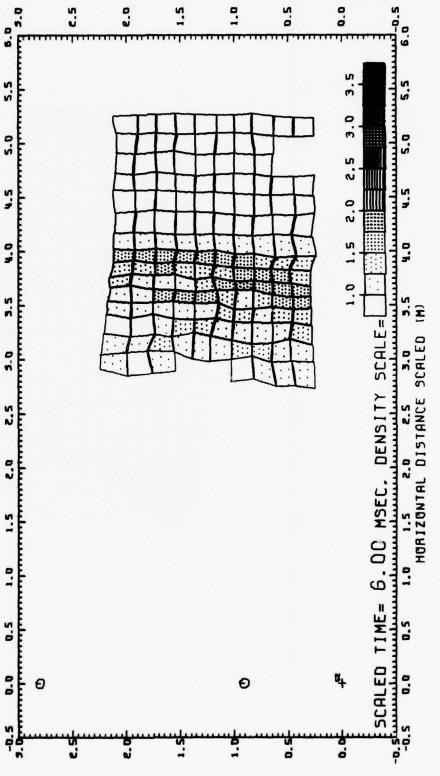
VERTICAL DISTANCE SCALED (M)

DENSITY FIELD, DIPOLE WEST/11

Fig. 13.1



VERTICAL DISTANCE SCALED IM)



DENSITY FIELD, DIPOLE WEST/11

VERTICAL DISTRNCE SCALED IM)

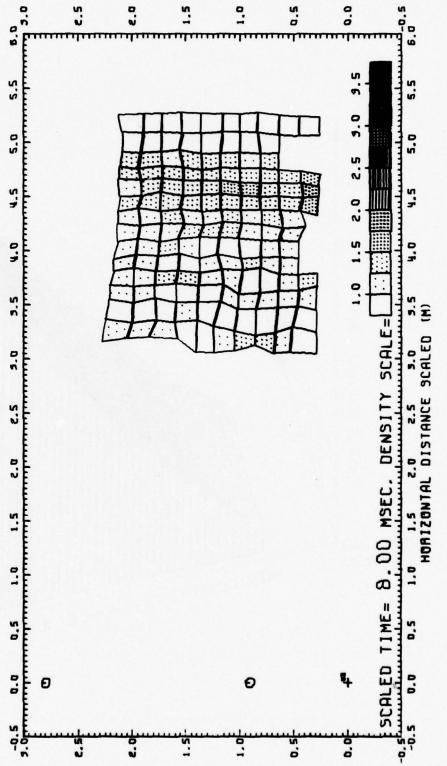


Fig. 13.4

DENSITY FIELD, DIPOLE WEST/11

VERTICAL DISTRNCE SCALED

(W) 59

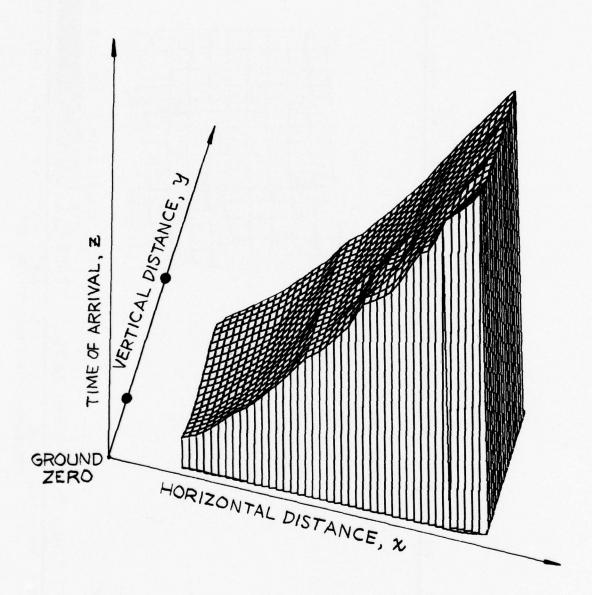
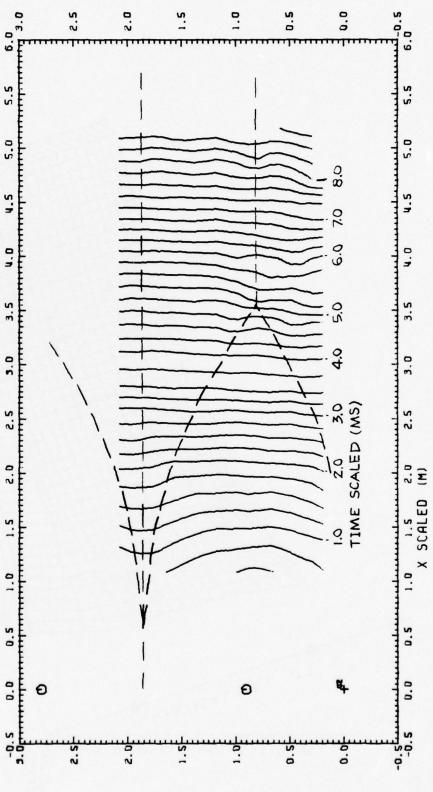
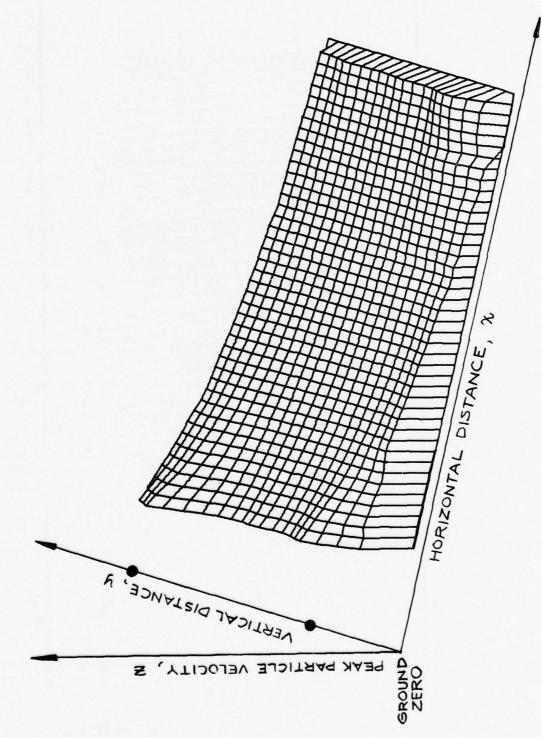


Fig. 14 Time-of-arrival surface

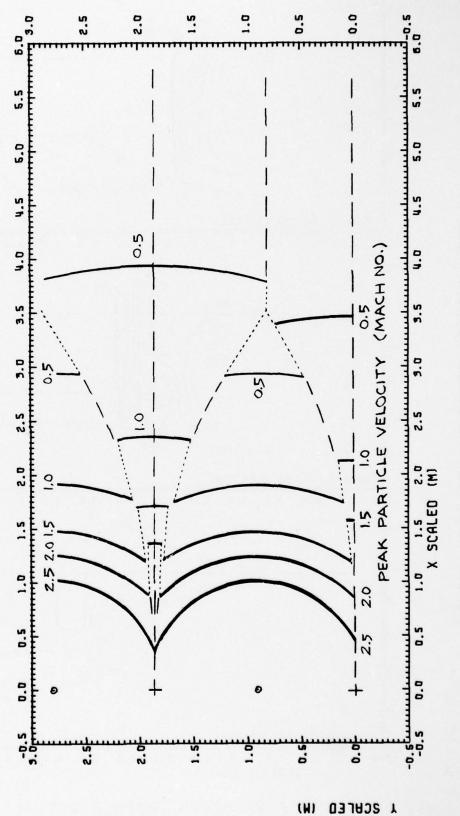


L SCHLED (M)

SHOCK FRONT SHAPES, DIPOLE WEST/11



A shock strength surface



SHOCK STRENGTH CONTOURS, DIPOLE WEST/11

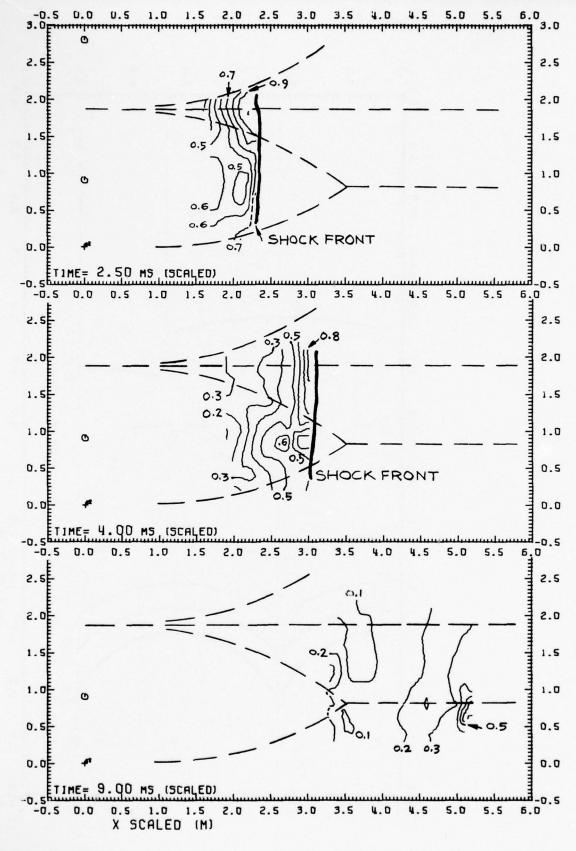


Fig. 18 PARTICLE VELOCITY, DIPOLE WEST/11

Ξ

SCALED

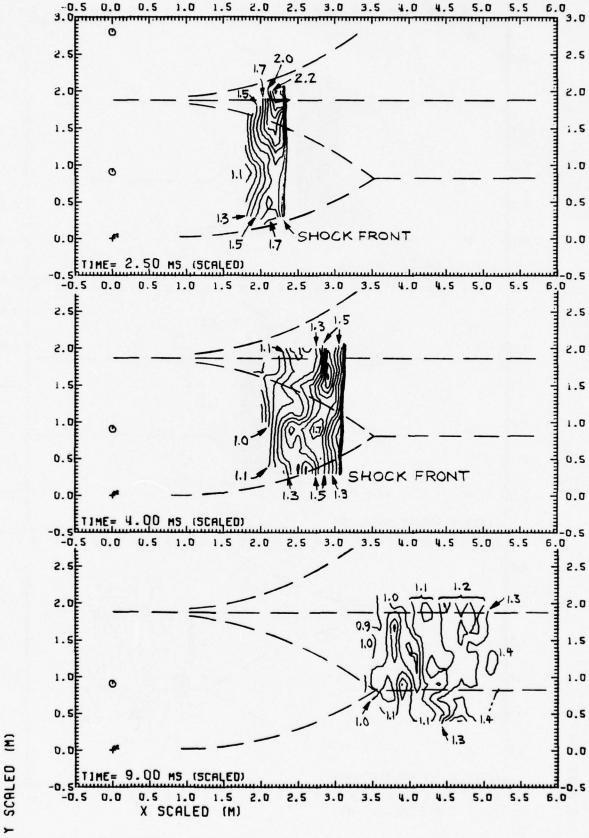


Fig. 19 DENSITY, DIPOLE WEST/11

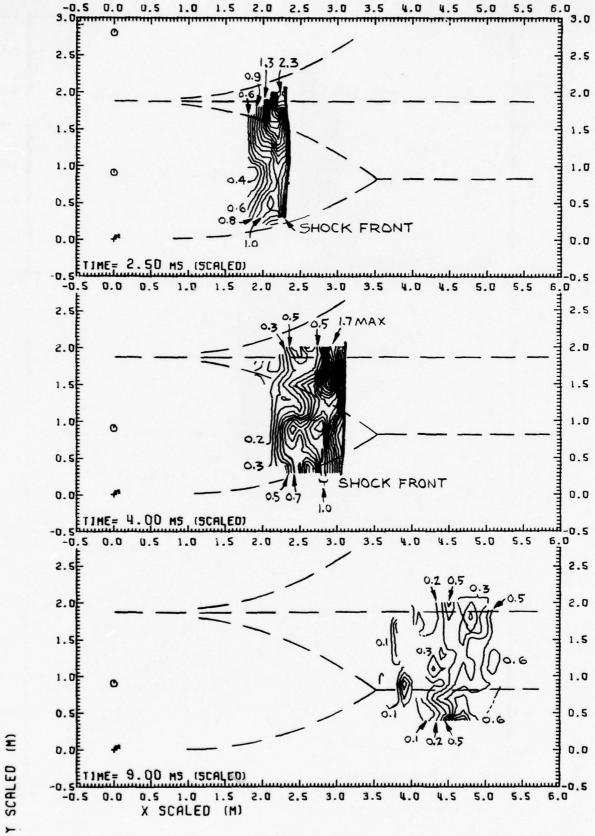


Fig. 20 HYDROSTATIC OVERPRESSURE, DIPOLE WEST/11

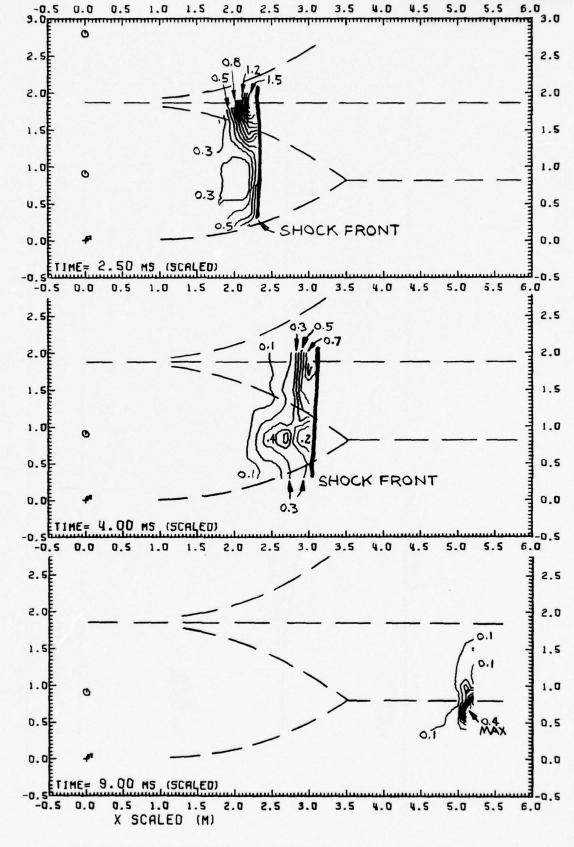
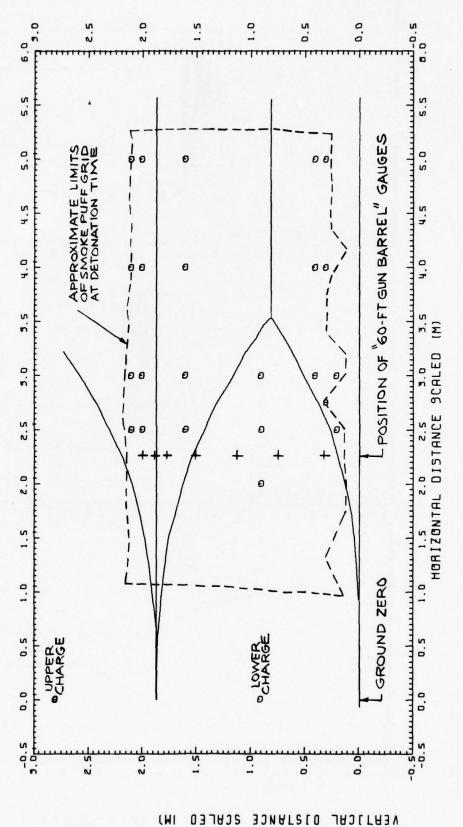


Fig. 21 DYNAMIC PRESSURE, DIPOLE WEST/11

SCALED



TIME HISTORY STATIONS, DIPOLE WEST/11

Fig. 22

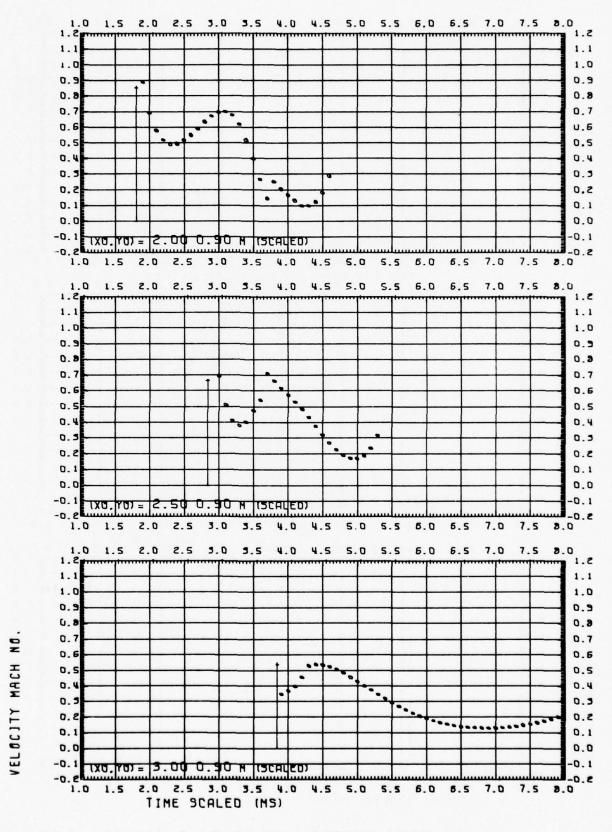


Fig. 23.1 PARTICLE VELOCITY, DIPOLE WEST/11

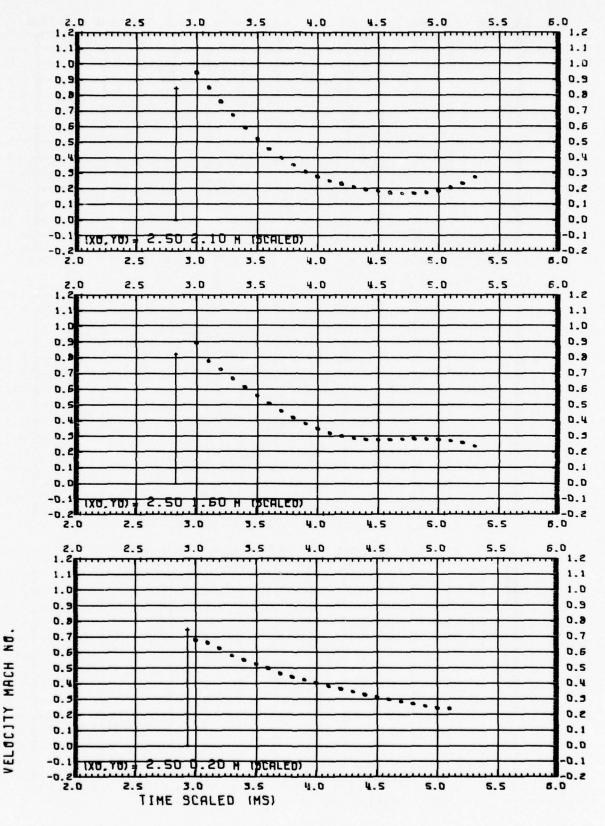


Fig. 23.2 PARTICLE VELOCITY, DIPOLE WEST/11

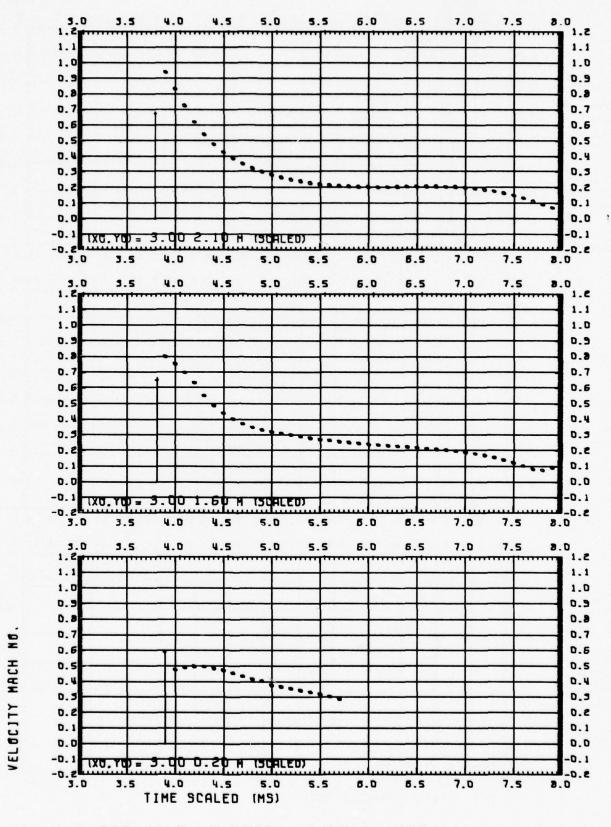


Fig. 23.3 PARTICLE VELOCITY, DIPOLE WEST/11

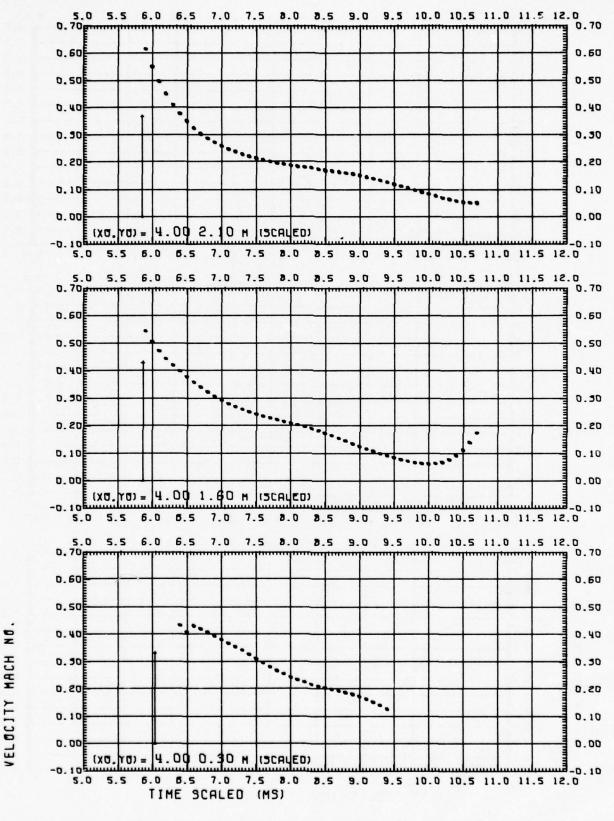
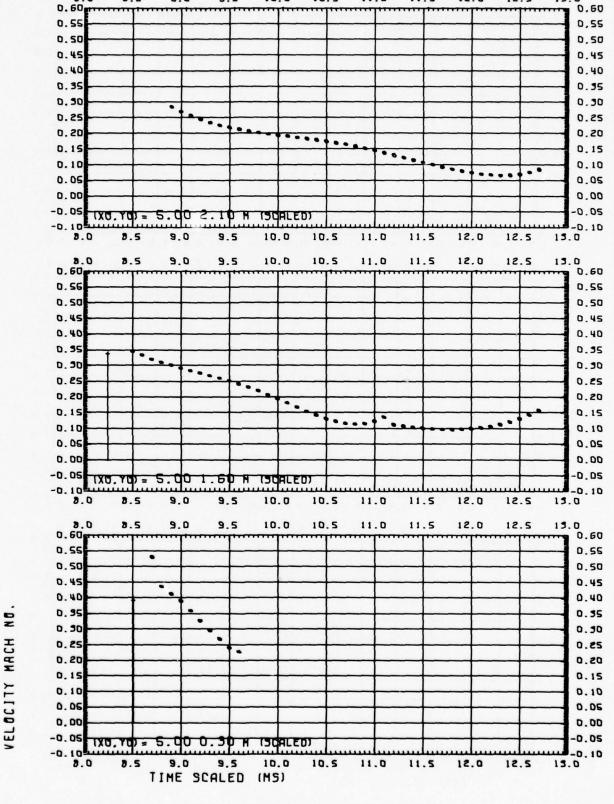


Fig. 23.4 PARTICLE VELOCITY, DIPOLE WEST/11



8.0

8.5

9.0

9.5

10.D

10.5

11.0

11.5

12.0

12.5

15.0

Fig. 23.5 PARTICLE VELOCITY, DIPOLE WEST/11

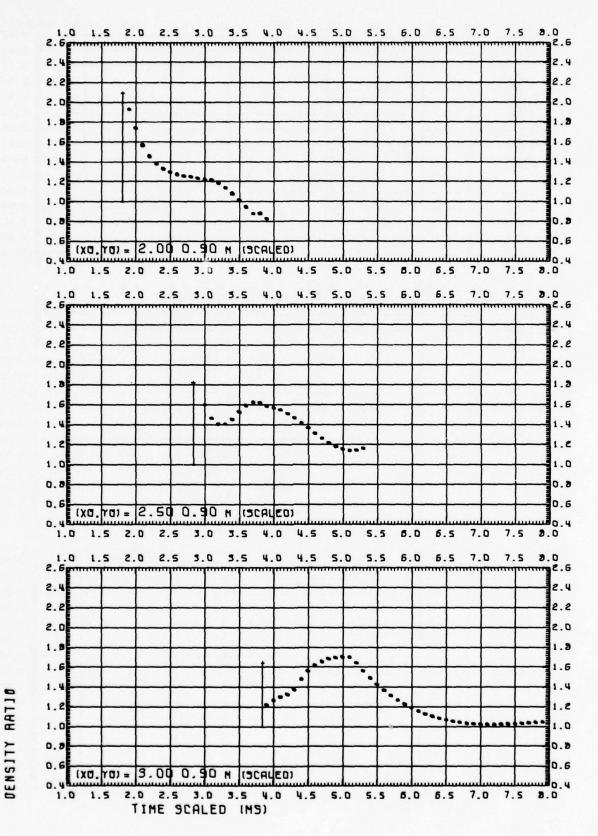


Fig. 24.1 DENSITY, DIPOLE WEST/11

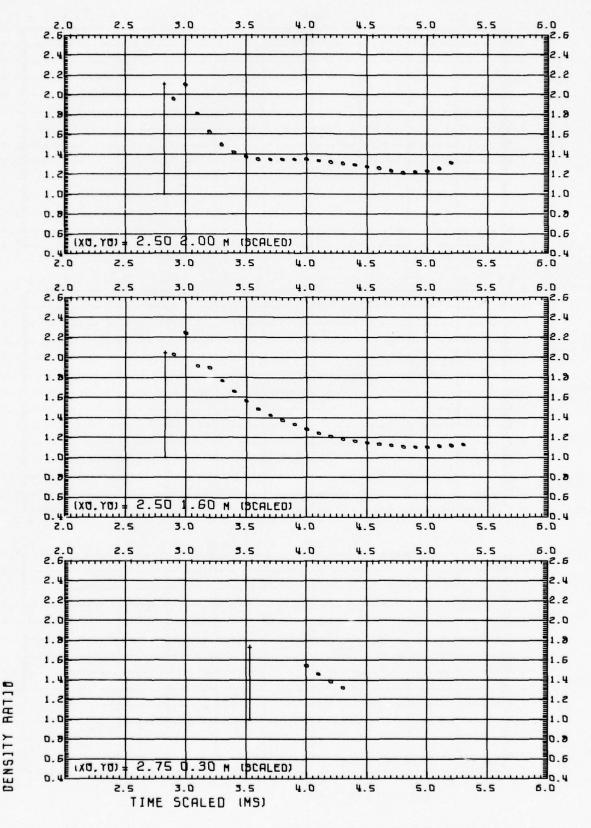


Fig. 24.2 DENSITY, DIPOLE WEST/11

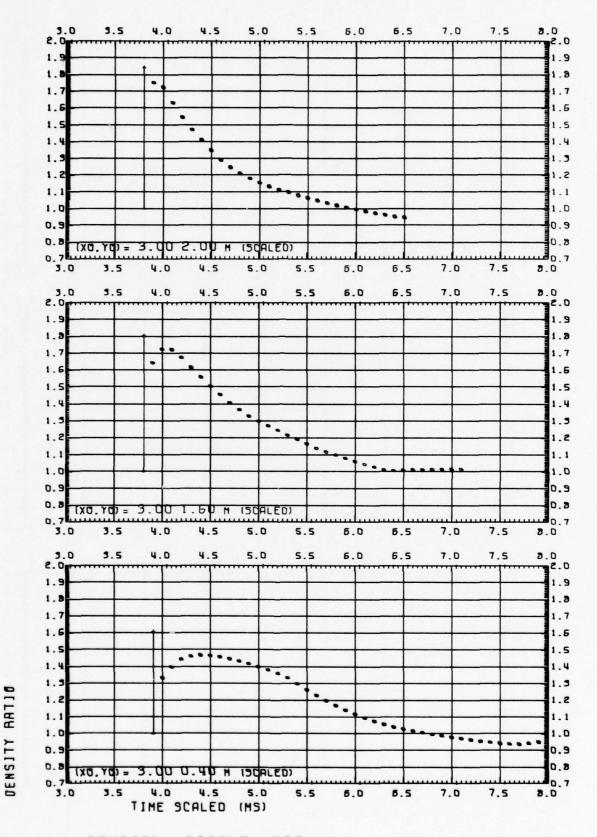


Fig. 24.3 DENSITY, DIPOLE WEST/11

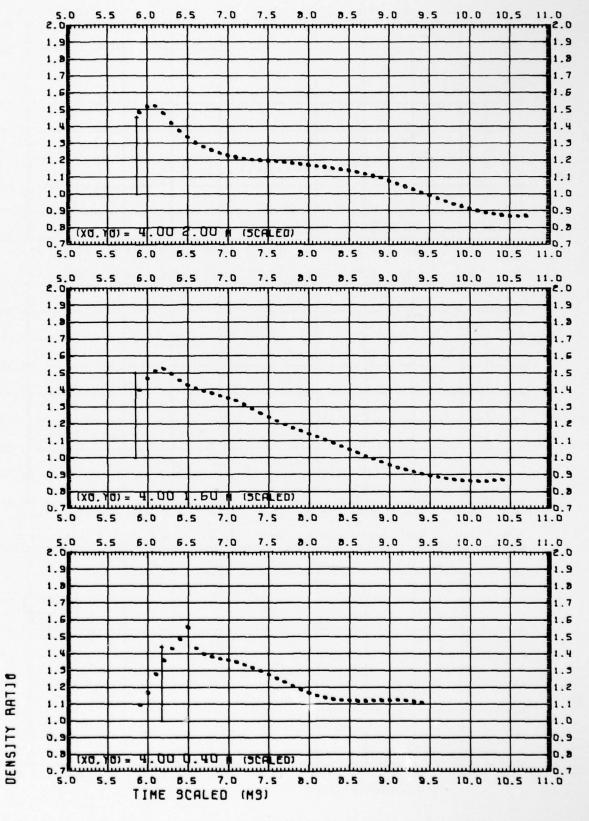
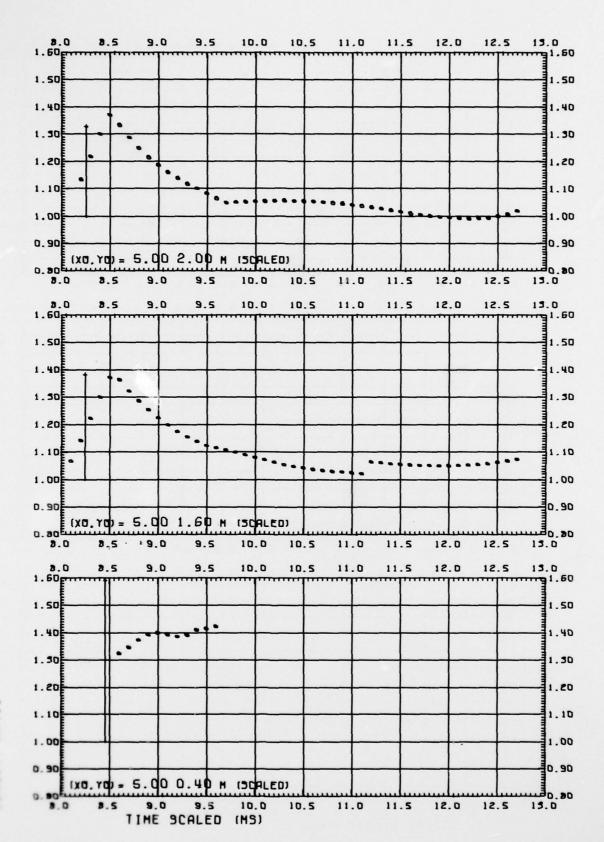


Fig. 24.4 DENSITY, DIPOLE WEST/11





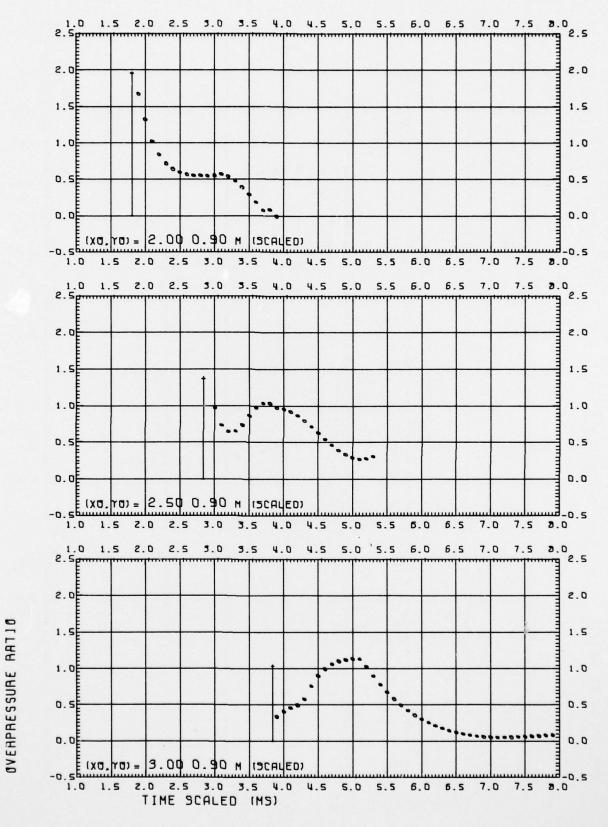


Fig. 25.1 HYDROSTATIC OVERPRESSURE, DIPOLE WEST/11

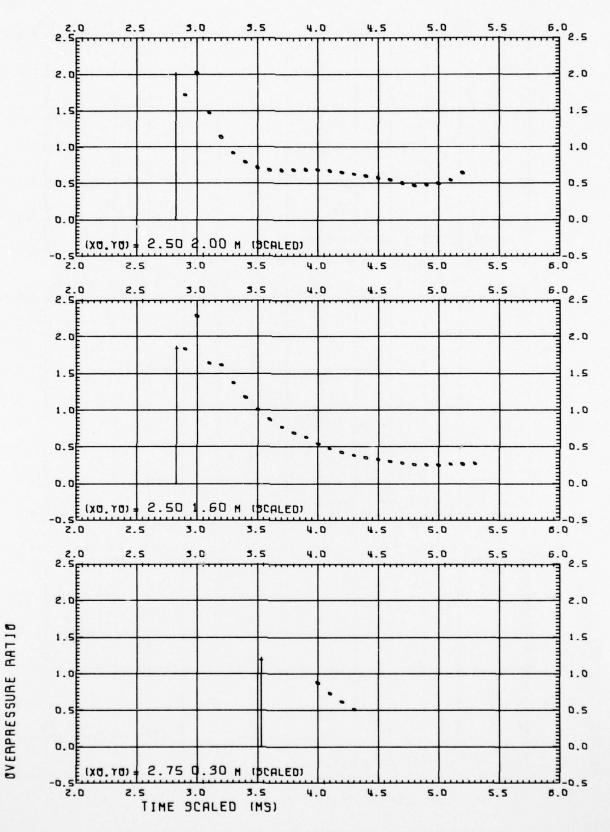


Fig. 25.2 HYDROSTATIC OVERPRESSURE, DIPOLE WEST/11

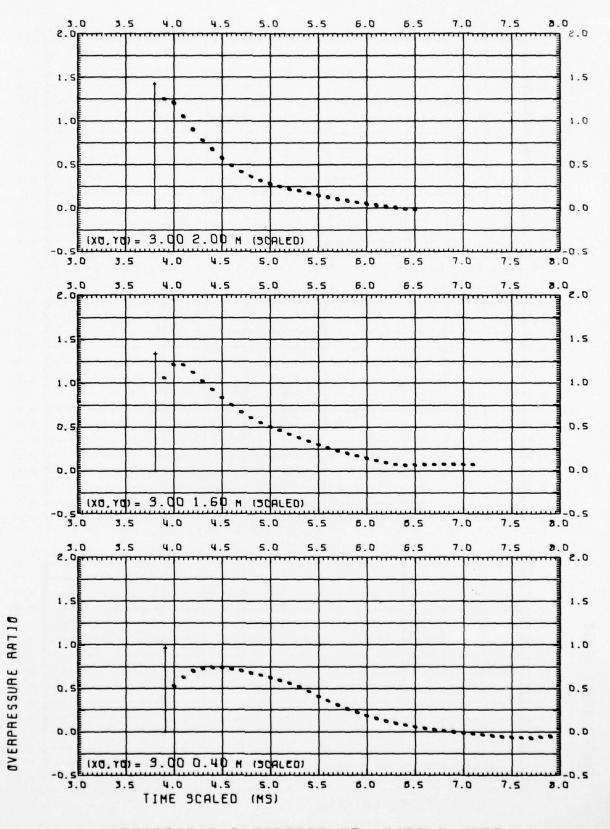


Fig. 25.3 HYDROSTATIC OVERPRESSURE, DIPOLE WEST/11

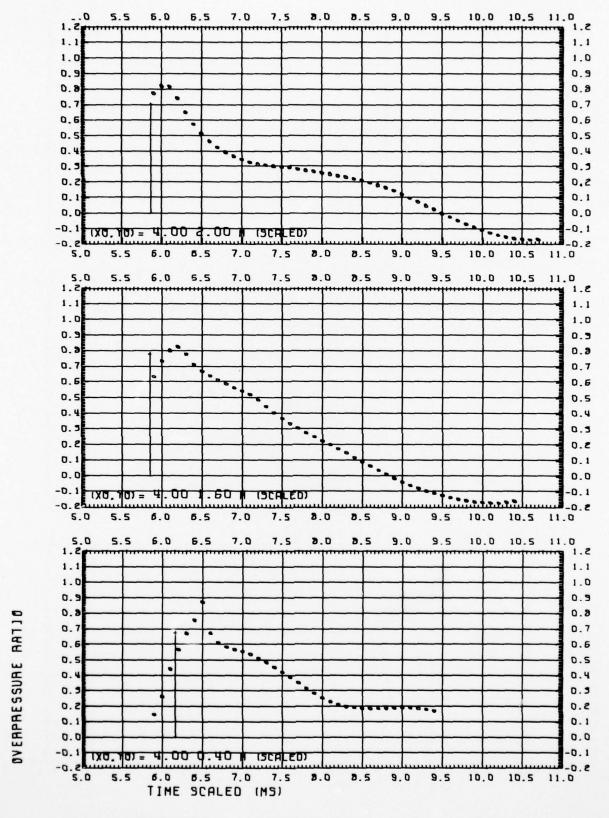


Fig. 25.4 HYDROSTATIC OVERPRESSURE, DIPOLE WEST/11

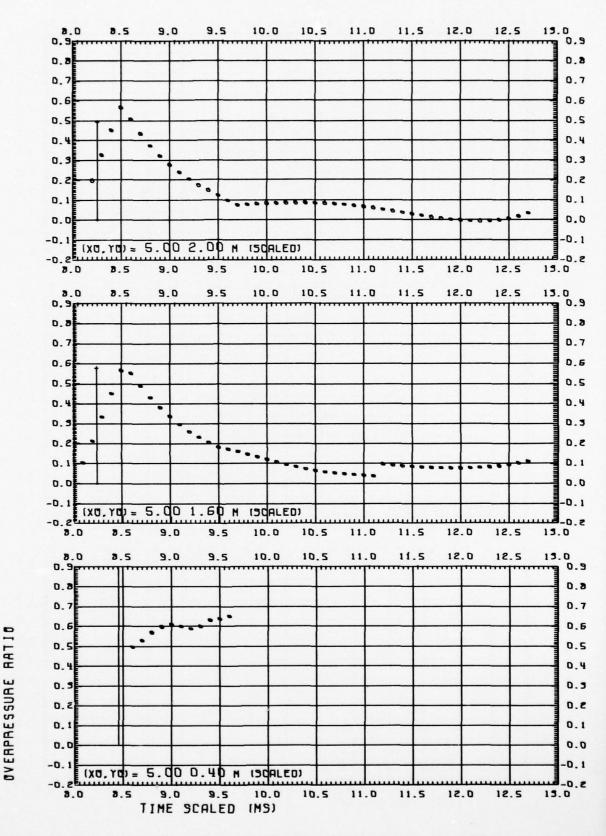


Fig. 25.5 HYDROSTATIC OVERPRESSURE, DIPOLE WEST/11

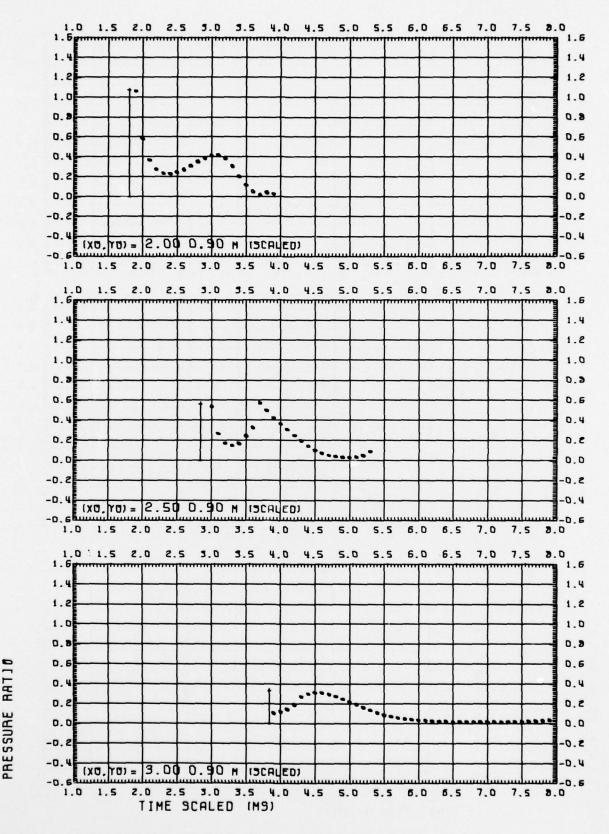


Fig. 26.1 DYNAMIC PRESSURE, DIPOLE WEST/11

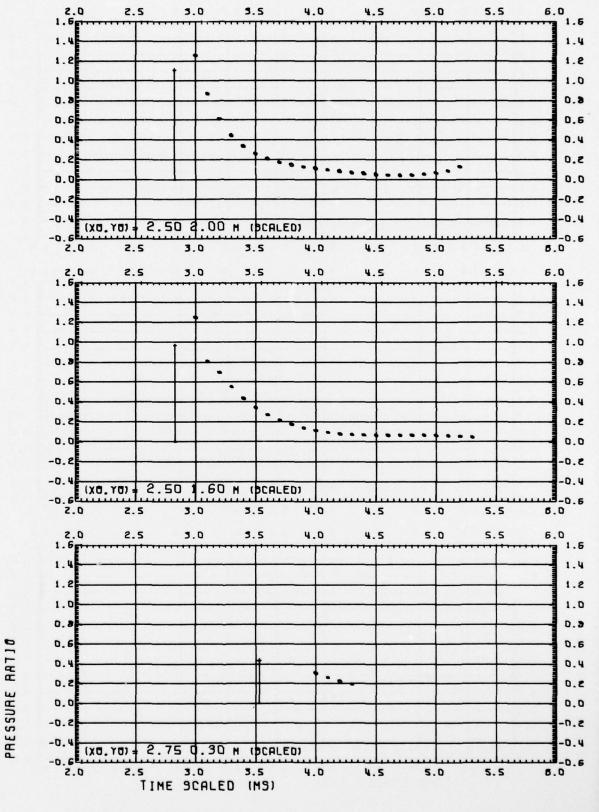


Fig. 26.2 DYNAMIC PRESSURE, DIPOLE WEST/11

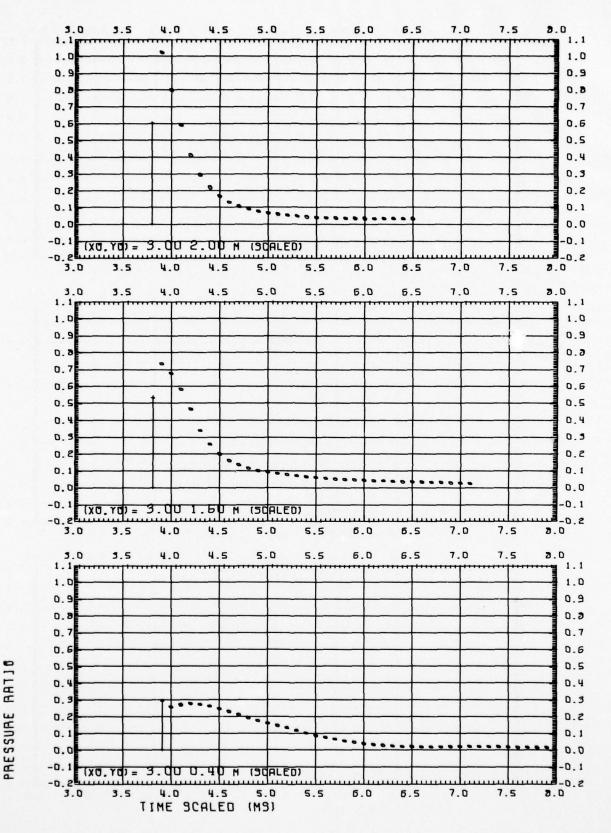


Fig. 26.3 DYNAMIC PRESSURE, DIPOLE WEST/11

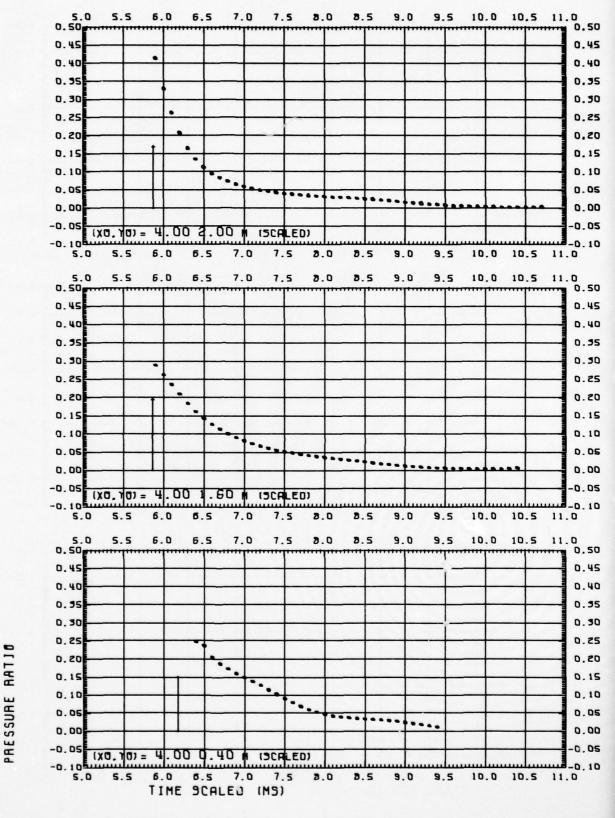


Fig. 26.4 DYNAMIC PRESSURE, DIPOLE WEST/11

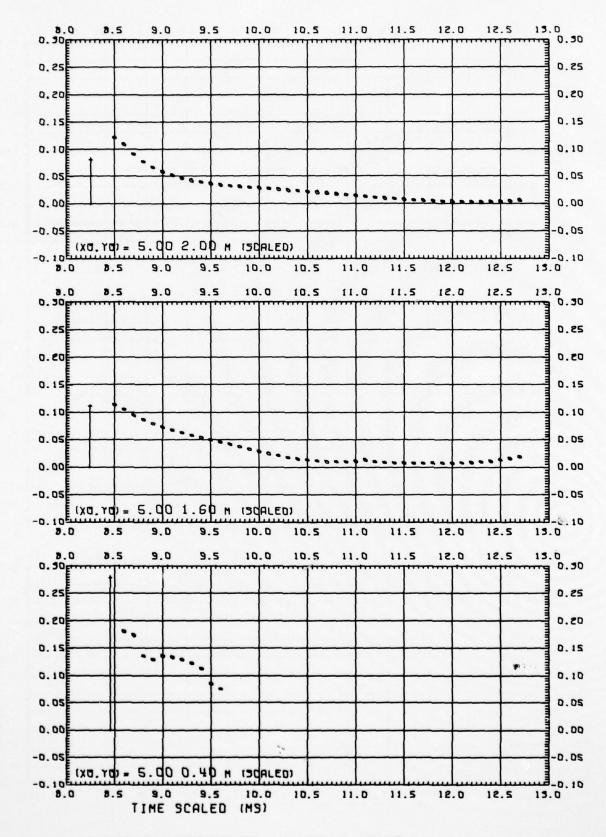


Fig. 26.5 DYNAMIC PRESSURE, DIPOLE WEST/11

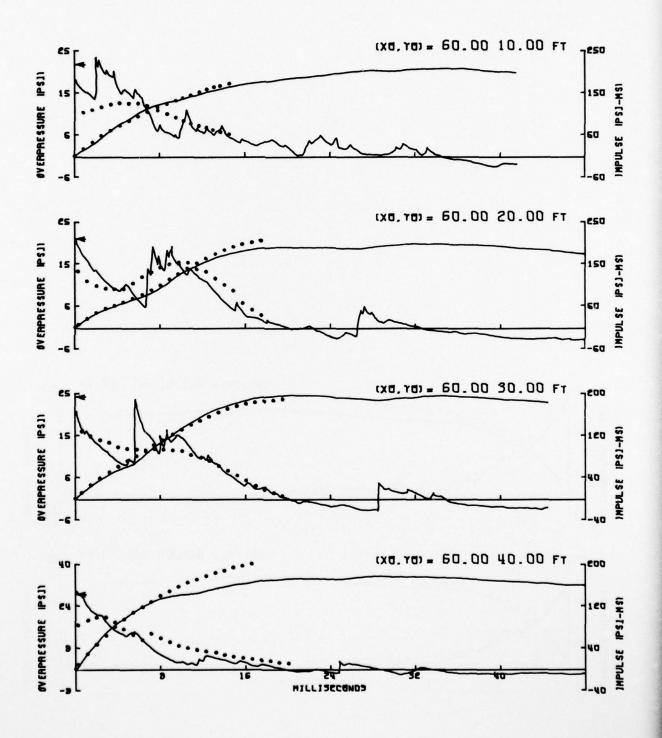


Fig. 27.1 DIPOLE WEST/11 HYDROSTATIC OVERPRESSURE

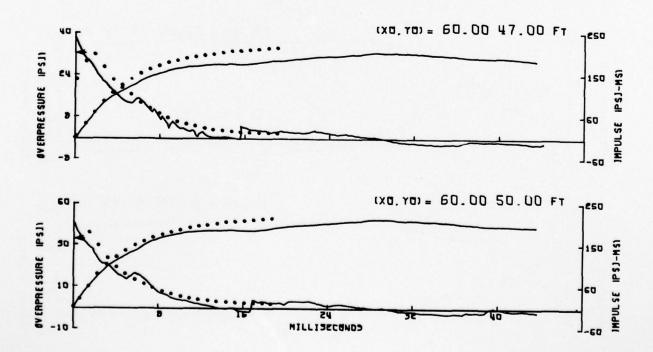
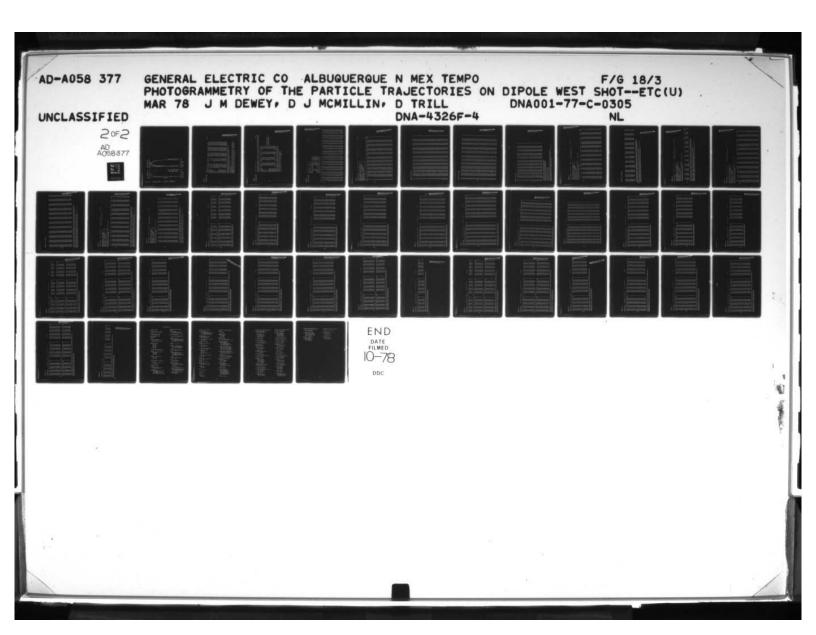


Fig. 27.2 DIPOLE WEST/11 HYDROSTATIC OVERPRESSURE
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Fig. 27.3 DIPOLE WEST/11 TOTAL PRESSURE



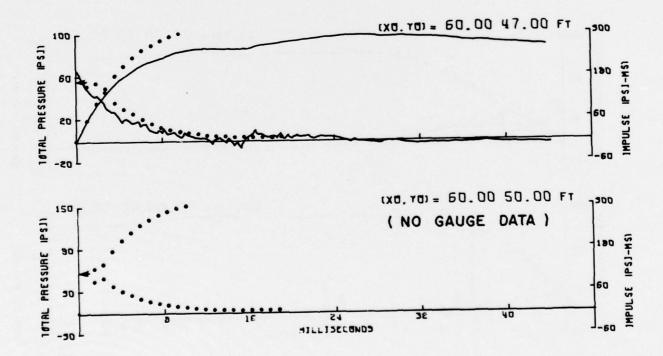


Fig. 27.4 DIPOLE WEST/11 TOTAL PRESSURE

DIPOLE WEST/11

SURVEY DATA LIST

TABLE

CJJRD. H	310.	316.32	310.32	340.29	390.41	313.75	340.64	343.82	2303.752	345.64	383.60	344.37	383.34	318.15	315.07	31 7.35	330.73	330.28	320.30	335.38	346.39	356.41	365.33	369.34	326.51	336.46	345.24	356.02	366.45	369.45	-
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DISTANCE	•	6.683		.60	0.763	.71		6.1	100.422	1.0	2.4	9.1	9.1	3.5	0.3	05.1	4.7	20.02		9.6	6	6.9	9.7	6.6	9.6	0.0	0	0.0	0.0	0.0	!
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SUNDRY DATA LIST

T= -2.4 DEG F, P= 13.7 PSI. RH= 60.0 %. SVP= 1.0 MM. W= 1040.0 LBS

T= -2.4 DEG F, P= 13.7 PSI. RH= 60.0 %. SVP= 1.0 MM. W= 1040.0 LBS

T= -2.4 DEG F, P= 13.7 PSI. RH= 60.0 %. SVP= 1.0 MM. W= 1049.0 LBS

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CALCULATED DISTANCE BETWELN G.ZERO AND G.ZERO C IS 0.723 FEET

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	VP 2A	-0.553	-0.203	164.0	0.206	
	VP 38	134.405	-0.343	-0.057	0.000	
	VP 38	-34.767	29.159	0.077	464.0	
	→ ?	34.855	-29.533	000.0	000	REFERENCE POINT PI
	* *	244.0	100.67	0.000	0.00	O TATELO
	300 W1	24.105	-27.047	0000	-0.012	REFERENCE POINT P3. PS
	300 ₩2	-22.242	-28.089	0.003	860.0	POINT P4
	1-20-10	89.172	-10.752	0.103	10.438	
	1-20.30	89.042	-0.734	0.255	-0-443	
	1-20.40	89.483	9, 157	0.358	-0.325	
	1-20.50	89.518	16.988	0.271	-0.244	
	1-20.53	69.494	8	0.220	-0.43B	
	1-30.50	97.878	18.984	10.003	10.445	
	1-30.53		22.071	•	0	
	AVERAGES			0.350	-0.504	
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20	100	6.368	7.071	• •	328	0.823	no	70	2000	3.7	
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25	11.064	3.873	7.675	1.370	0.480	0.000	1.194	10.369	2500	1.435	
25	4 . 5	17,309	13.998	•	14	1.629	0	40	1.085	5	
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23	14.166	11.051	12.194	•	30	1.419	. 10	40	0.924	8.0	
30	14.175	9.532	12.194	•	200	1.419	N:	500	1.258	17.	
32	14.132	7.012	12.194		300	1.419	+ 10	0	040	2.5	
33	14.379	5.609	12.194	•	69	1.419	m	0	1.359	.79	
45	13.988	3.759	12.194	•	40	1.419	n	.34	0.902	.73	
35	14.034	2.397	13.096	•	500	1.524		500	1.000	9.84	
37	16.403	17.255	16.999	• •	130	1.978	- m	17	40000		
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65	16.540	14.297	16.999	•	1 2	826.1	200	.17	168.0	30.	
4	16:391	11.310	15.799	• •	40	1.439	m	000	0.343	000	
42	16.475	9.374	15.799	•	22	1.839	~	.07	0.744	000	
43	16.462	8.491	15.799	•	0 5	1.839	0.	6	1.001	*00	
4 4	16.070	5.371	5.799	٠.	99	1.839	-	7 4	1.152	50	
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TABLE

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DENSITY	4 4 M E. O.	3.505 3.612 3.8612 3.860 8.860	33.11.0 3.01.7 3.01.7 2.958	22.22.22.22.23.23.23.23.23.23.23.23.23.2	44 42 8 8 4 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6	000000 TTTTTTTTTTTTTTTTTTTTTTTTTTTTTTT
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PARSSURE KPA	1105.271	7782-139 722-139 722-139 590-139	94444 9465. 94444 9465. 9465. 9465. 9465. 9465.	4700 4700 4700 6700	257.845 237.845 237.836 217.270 217.270 189.237 171.137	1711.137 1711.137 1711.137 1711.137 1711.137 167.044 124.724 124.724 124.724
PRESSURE RATIO			0.004444 0.0007 0.0007 0.0007 0.0007		% % % % % % % % % % % % % % % % % % %	11.1.1.1.1.1.1.1.1.1.1.1.1.1.1.1.1.1.1
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8-0-8 8-15-8	8.284 8.457 8.873	8,978 9,347 9,351 10,072	11.047 11.047 11.0585 11.0585	12.2.2.2.4.4.4.4.4.4.4.4.4.4.4.4.4.4.4.4	44444 2000 2000 2000 2000 2000 2000 200	10.000 10.000 10.000 10.000 10.000 10.000 10.000 10.000
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TABLE 5.1

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SHOCK FRONT DATA COMPUTED FROM PARTICLE TRAJECTORY TIMES OF ARRIVAL RFIT=A +5*F+C*LCS(1+T)+D*SQRT(LOG(1+T)) USING WEIGHT=INVERSE OF RADIUS SQUARED

1.794	1.627	1.627	1.627	1.627	1.619	1.503	1.503	1.503	1.508	1.508	1.476	1.471	1.471	1.427	1.427	1.397				1
0.647	020.0	0.556	0.526	0.526	0.520	0.437	0.437	0.437	0.437	0.437	0.412	0.400	0.408	0.374	0.374	0.350		0. V E D •		
124.724	94.700	94.705	94.705	94.705	93,373	74.715	74.715	74.715	74.713	74.715	69.444	68.733	68.754	61.610	61.610	56.880		18 SPHCAICAL SPECT OSSER DENSITY D.		DEGREES CELSIUS)
1.322	1.004	1.004	1.004	1.004	0.66.0	0.792	0.792	0.732	0.792	0.792	0.736	0.729	6.729	0.653	0.653	0.003		SHAPE IN KILDP AMBIENT	GCOND	
1.401	1.364	1.364	1.364	1.364	1.300	1.296	1.296	1.296	1.296	1.296	1.277	1.275	1.275	1 : 249	1.249	1.232		SA SHOCK FROM THE CAMPAND TO THE CAMPAND THE TOTHER THE THE THE CHAPTER THE CH	92 METERS/S (W/WUJ # (PO/	AMBIENT (T
2.298	2.581	2.331	2.581	2.541	2.597	2.378	2.378	2.378	2.373	2.478	2.984	2.399	2.999	3.177	3.177	3,322		UTED ASSUMI TS. RELATIV K UVERPRESS RATIJ. REL	65 CO 3 40 2	HERE WHERE COLOND SO ARE AMBIENT (TO= 15) AS ANTIOS. ARE INVARIANT UNDER SCALING.
2.361	2.952	2.952	2.952	2.952	2.957	3.610	3.010	3.610	3.010	3.610	3.852	3.387	3.847	4.300	4.300	4.645		SITION COMPINATOR WASHINGTON TO THE SEED AS A A A A A A A A A A A A A A A A A A	CO)/S, *HER EO 3Y S= CU	RE WHERE CO
0.277	0.012	0.102	0.243	0.249	0.118	0.439	0.184	0.537	0.676	0.634	-0.335	-0.300	-0.322	-0.330	-0.327	-0.491	0.280 RMS	RADIAL PUFF PUSITION COMPUTED ASSUMING SHICK FROM ARE EXPRESSED IN MACH UNITS, RELATIVE TO FHE AMBIE RATIO (PMAX-P)/P, AND PEAK OVERPRESSUME (PMAX-P) DENSITY IS EXPRESSED AS A RATIO, RELATIVE TO THE A	JETANCE DIVIDED 3Y	IN ATMOSPHE EXPRESSED A
18.552	20.439	20.97.9	20,439	20.439	20. 363	23.239	23.230	23.238	23.238	23.238	24.093	24.214	24.214	25.047	25.547	26,919		(010:11	TI ME MULTI	O CHARGE WO
18.276	20.426	20, 736	20.595	20.550	20. 450	22.799	23.054	22.701	22.562	22.554	24.428	24.513	24.536	25.577	25.974	27.710		DF ARRIVAL AND R IS PARTICLE VELOCITIES S PEAK DVERPARSSUR MADIENT PRESSURE.	SCALED TIME= DBSEKVED TIME MULTI AND SCALED DISTANCE= DBSERVED DI MACHE POSE DOI: 325 KT. DBASCALS.	TESSUAE. A
20.293	25.370	25.370	25 - 276	25.370	25.008	31.0025	31.025	31.025	31.025	31.025	33.103	33.400	33.400	36.955	36.955	39.912		SHUCK AND PRESSURE IS	SCALED TIME	SCALED EVEN

QUALI: ED TO I	LA LIMOTZGIA					
ED TO			DENSITY	2.904 2.830 2.627 2.423 2.317		
50			PARTICLE	1.360 1.0822 1.0822 1.082	tsvē. RVĒU.	:
R2 /A730120			PAESSURE APA	406.278 381.247 313.657 263.169 236.264	ISSPHEAICA DSSEED USSE DENSITY D.	snishso saa
R CHARGE			PRESSURE RATIO	4.306 4.041 3.378 2.790 2.504	TAL PUFF PUSITION COMPUTED ASSUMING SHOCK FRONT SHAPE IS SPHERICALIVE. TO THE AMBIENT SOUND SPHERICALIVE. TO THE AMBIENT SOUND SPHERICALIVE. TO CPMAX-P) IN KILOPASCALS DESERVED SITY IS EXPRESSED AS A RATIO, RELATIVE TO THE AMBIENT DENSITY O.	LICO BY (C/CO)/S, *HERE CO= 340,292 MCTERS/SFCOND TANCE DIVIDED BY S= CUBL ROOT 3F (*/WO)FFPS/SFCOND *, *O, AND P ARE DEFINED ABOVE.) IN ATMOSPHERE #HERE CO AND PO ARE AMBIENT (TO= 15 DEGREES CELSIUS). EXPRESSED AS RATIOS, ARE INVARIANT UNDER SCALING.
SMJKE PUFF GRID 1220 PRIMARY FRONT FRJM UPPER CHARGE		٩ ^٢	SHJCK VELUCITY	2-160 2-113 1-374 1-341	ING SHOCK F VECTO THE A SORE (PMAX-	292 MCTERS/ (W/WC)*(PO E AMBIENT (
SMOKE PUFF GRID 1220 PRIMARY FRONT FROM U		ES OF ARRIVAL	R-SCAL METERS	1.278 1.355 1.599 1.916 2.131	PUTED ASSUMITS. RELATION RES	CGE 340F NEC CGD 340F NED ABOV C.) O AND PO AR
30.	0 V	ECTORY TIM	T-SCAL MSCC	0.682 0.788 1.139 1.629 1.973	SITION COM	CO)/S, whe con S ARE DEFI
WF5/295	SCALS CELSIUS SCALS SCALS MATERS/SECD SCALS MATERS/SECD	ARTICLE TRAJECTORY TIMES FIXED AT 1.0 NUARED	DIFFERENCE	0.038 0.028 0.038 0.042 0.042	AL PUFF PU EXPRESSED (PMAX-P)	ANCE DIVID
01P3L5 #25T/1	1	TED FROM PAR TANDIUS SOU	METERS	10.320 10.940 12.906 15.469	RAND EATED	SERVED DIST PASCALS. C CHARGE WO DENSITY. E
	SCUAME VALUE OF THE VALUE OF TH	*LOG(ALPHA- =INVERSE OF	R-08S MET ERS	10.288 10.998 12.878 15.434 17.244	ARPIVAL APERA PERA OVERPENT PRES	OBSERVED ISTANCE DE STANDARD ESSURE, AND
SHUCK FRONT DATA	AMJIENT TEMPERATURE T= -19-11 DEGREES CELSIUS ANGIENT PRESSURE P= 94.34 KILUPESCÂLS CELSIUS ANGIENT PRESSURE P= 94.34 KILUPESCÂLS CELSIUS PELATIVE HUMIOITY RH= 60.00 KILUPESCALS VAPOUT PERSSURE P= 0.00 KILUPESCALS CHARGE AEIGHT W= 7.31 MÉTEKS PARSON CHARGE AEIGHT W= 7.31 MÉTEKS PERSOND CHARGE HIGHT PS 7.31 MÉTEKS PERSON SCALING FACTUS SCALING FACTUS SCALING FACTUS SCALING FACTUS PS 9.0730 SCALING TU CHARGE WEIGHT WO= 1.00 KILUPESCANS	SHOCK FRONT DATA COMPUTED FROM PARAFIT=AFB#T+C#LDG(ALPHA+T), ALPHA LUSING WEIGHT=INVERSE OF RADIUS SOU	7-083 MSEC	5.863 6.769 9.785 13.998 16.999	T IS TIME OF ARRIVAL AND K IS SHOCK AND PARTICLE VELOCITIES PAESSURE IS PEAK OVERPRESSURE WHERE P IS AMBIENT PRESSURE	SCALED TIME= OBSERVED TIME MULTIPHE AND SCALED DISTANCE= DBSERVED DISTANCE= BOSERVED DISTANDARD CHARGE WOLLD EVENT= STANDARD CHARGE WOLLD CHARGE WOLLD EVENT= STANDARD DENSITY.

R4 /A730120

MACH STEM GULOW INTERACTION PLANE

SMUKE PUFF GRID 1220

30.

DIPOLE #EST/11 #F5/295

SHUCK FRONT DATA

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AB
H

## 19-11 DEGREES CELSIUS 0-0.3 KIECPASCALS 0-0.3 KIECPASCALS	######################################
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•		4.6	. 57	3	.28	. 35	0.016	N	20
•	•	4.5	.57	3.	.28	. 85	-0.822	'n	44
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RADIAL PUFF POSITION COMPUTED ASSUMING SHOCK FRONI SHARB IS SPUTAICALLABLE.
ARE EXPRESSED IN MACH UNITS. RELATIVE TO THE AMBIENT SOUND SPLET (ABBVE).
RATIO (PMAX-F)/P, AND PEAK UNLYPRESSOUR (PMAX-P) IN KILOPASCALS USSERVED,
DENSITY IS EXPRESSED AS A RATIO, RELATIVE TO THE AMBIENT DENSITY O. SHIDGE AND R IS SHIDGE AND PARTICE THE PRESSORE IS A WELL THE SSURE.

SCALED TIME= UBSERVED TIME MULTIPLIED BY (C/COD//S, WHLRE CO= 340,292 METERS/SFGOND AND SCALED DISTANCE OBSERVED DISTANCE DIVIDED BY SE CULL ROOT OF (W, WU) *(PG/P); OBSERVED DISTANCE DISTANCE DISTANCE DISTANCE DISTANCE DISTANCE DISTANCED BY OF SCALED EVENTE STANDARD CHARGE WITH THROSPHERE WHERE CO AND PO ARE AMBIENT (TO= 15 VELOCITY, PRESSURE, AND DENSITY, EXPARSSED AS NATIOS, ARE INVARIANT UNDER SCALING.

DEGREES

THIS PAGE IS BEST QUALITY PRACTICABLE FROM COPY FURNISHED TO DDC

10 40%	30	14	32	20	מי	36	113	52	24	10	01	24	40.	.	75	75	42	39	*	12	46	40	74	14	58	57	45	63		
	2.530																													
V=LGCITY 1.692	1.327	0.950	0.021	0.085	0.641	140.0	00.0	0.544	145.0	0.500	0.200	0.471	74.0	174.0	0.411	0.411	0.335	0.384	0.363	0.362	0.348	0.348	3332	0.332	0.319	0.318	0.308	0.307		
545.035	363.505	211.703	174.250	135,093	123.223	123,223	109.850	99.076	98.200	90.236	90.230	82.275	82.273	75.169	69.277	69.277	63.984	63.641	59.540	59.256	56.539	50.539	53.432	53.432	50.923	50.723	48.378	48.703		
	3.087																													
VEL3CITY 2.440	2.074	1.710	1.500	1.492	1.456	1.456	1.410	1.378	1.376	1.349	1.349	1.322	1.322	1.297	1.276	1.276	1.253	1.256	1.241	1 . 240	1.230	1.230	1.219	1.219	1.209	1.209	1.202	107.1		
MET_KS 1.443	2.302	24.000	2.579	2.925	3.063	3.063	3.269	3.428	3.444	3.604	3.504	3.790	3.790	3.588	4.133	4.183	4.388	4.402	4.590	4.604	4.747	4.747	4.929	676.4	2.096	5.110	5.248	5.262		
45 EC	1.664	2.535	2.987	3.610	3.887	3.887	4.300	4.645	4.679	5.022	5.022	5.434	25.00	5.378	6.322	6.322	6.198	6.332	7.273	7.307	7.645	7.645	8.044	8.084	8.499	8.522	8.858			
0.055	10.056	C.278	-0.126	0.686	0.028	-3.188	-0.070	-0.214	-0.208	-0.153	-0.072	-0.258	817.0	-0.032	-3.012	-0.027	0.117	0.194	0.133	0.141	0.227	0.032	0.150	0.015	0000-	-0.010	-0.136	-0.038	0.201 AMS	
METERS 11.646	14.446	14.922	20.971	23.613	24.732	24 . 732	26.326	27.677	27.807	29.092	29.092	30.00	20.000	32.199	33.766	33.766	35.424	35.541	37.054	37.170	34.319	33, 319	39.795	39.195	41.144	41.255	42.363	45.479		
MCTERS 11.591	14.502	18.643	21.097	22.927	24.704	24.919	26.371	27, 391	28.015	29.275	29.164	30.369	30.018	32, 231	33,779	33,794	35.307	35.348	35. 921	37.029	38.091	38.236	39.646	39. 730	41.144	41.255	45.554	45.567		
MSEC 10.333	14.298	21.788	25.658	31.025	33.400	33.400	30.955	39.912	40.208	43.158	43.158	46.692	260.00	50.512	54.323	54.323	53.416	53.708	65.438	65.183	20.00	65.698	69.471	07.471	12.945	73.234	76.122	10.410		

RS /A740120

PLANG

MACH STEM ABOVE INTERACTION GRIU 1220

SMUKE PUFF

WF5/295

#EST/11

DIPOLE

DATA

SHOCK FRONT

5.4

TABLE

ARRIVAL

OF

TIMES

SHOCK FRONT DATA CCMPUTED FROM PARTICLE TRAJECTORY STATTA-BETT-1) TRAJECTORY STATTA-BETT-1) TRAJECTORY STATTA-BETT-1) TRAJECTORY STATTA-BETT-1 SQUARED USING WEIGHT-BINVERSE OF PADIOS SQUARED

AMBIENT TEMPERATURE T= -19-11 DEGREES CELSIUS
AMBIENT PRESSURE P= 94-34 KILCHSCALS
ARILTIVE MUNIDITY KATE 60.6 PEA CENT
VAPOUR PRESSURE VP= 0.63 KILCHSCALS
ANBIENT SPEED F SOUNC C= 319-689 METERS/SECOND
CHANGE ASIDATINE 7-31 METERS
SEPARATION 42 HS= 7-31 METERS
SEPARATION 42 HS= 7-64 METERS
SCALING FACTOR S= 8-0730
1.0 KILCHSRAS

1 IS TIME OF ARTIVAL AND R IS RADIAL POFF POSITION COMPUTED ASSUMING SHOCK FRONT SHABE IS SPHERICALED TO THE ANAIENT SOCIAL SPEER IS SPEER CABBOVE. PRESSURE ARTIO (PMAX—P) PP. AND PARTS OF CAPARESSURE RATIO (PMAX—P) PP. AND PARTS OF CAPARESSURE RATIO (PMAX—P) PP. AND PARTS OF CAPARESSURE. FATIO (PMAX—P) PP. AND PARTS OF CAPARESSURE. DENSITY DECEMBER PIS AMBIENT PRESSURE. SCA_ED TIME= 08SERVED TIME MULTIPLIED BY (C/CO)/S, WHERE CU= 340,292 METERS/SECOND
AND SCALED DISTANCE= 03SERVED DISTANCE DIVIDED BY S= CUBE ROLT OF (W/WO)*(PO/P),
MHERE PO= 161,325 KILDPASCALES, (W, *0, *0) ND P ARE DEFINED ADDCE.
SCALED CVENT= STANDARD CHARGE *0 IN ATMOSPHERE WHERE OF AND PO ARE AMBIENT (TO= 15 DEGREES VELOSITY, PRESSURE, AND DENSITY, EXPRESSED AS AATIOS, ARE INVARIANT UNDER SCALING.

CELSIUS).

THIS PAGE IS BEST QUALITY PRACTICABLE

T IS TIVE OF ARTIVAL AND A 18 RADIAL FUEF POSITION COMPUTED ASSUMING SHOCK FRONT SHADE. IS SPHEATICALLY AND ARTICLE AND ARTICLE ASSUMING SHEED CAND PRESSURE.

SHADE IS PRESENTED AND SHEED CANDERS OF AND AND AND ARTICLE AND ARTICLES ASSENCE.

IN AND ARTICLE AND ARTICLES AND ARTI

			DENSI LY	2.4.37	2.133	1.973	1.730	1.730	1.730	7 000		1.627	1.614	1.505	1.553	1000	1.520	1.521	C. + + 1	1.445	1.479	1.471	1.471	004.1	44.	1.437	1.43	1.424	1.419	204	1.402	1.347	062.1	100	1.383	
50			PAR TICLE VELUCITY		0.920																															
R3 /A780120			PRESSURE ARA	20.3.714	202.473	160.603	114.004	114.004	114.004	101.000	100.181	94.05+	92.409	83.074	83.874	70.473	77.001	76.530	72.545	72.045	010.00	69.707	64.707	65.65	000000	63.203	65.660	61.233	42 4.00	30.00	57.762	50.934	20.00	500.00	5000	
SURFACE			PRESSURE RATIO	2.736	2.132	1.702	1.209	1.239	1.209	000	1.000	1.003	000000	6.830	0.000	0.832	0.823	4.8.0	602.0	602.0	147.0	0.728	0.728	2000	0.032	0.070	000	540.0	2000	0.613	0.012	6000	2000	585.0	00.00	
SED 1220		ه ر	SHOCK	1.444	1.094	D	1.427	1.427	1.427	200	1.332	1.364	1.350	1.327	1.327	1.309	1.390	1.303	1 .298	1.233	1.279	1.274	1.274	1 . 202	1.259	1.255	1.254	1 - 2 4 7	242	1.237	1.235	1.232	1.230	1.227	1.424	
SMUKE PUFF GF		ES OF ANTIVAL	A-SCAL METIFIS	1.342	2.233	2.552	3.013	3.013	3.168	3.211	3.244	3.350	3.404	3.023	3,043	3.792	3.36.5	7.00	4.018	4.0.4 9.0.8	4.137	4.197	4.197	4.387	4.415	610.4	0 0 0 0 0	4.500	4.7.9	4.105	4.361	20.01	501.5	5.201	5.2.9	
30.	2 NO 8	TAAJECTORY TIMES	T-SCAL	2.013	2.431	356.5	3.921	3.921	3.921	4 3 13	404.4	4. 545	4000	5.228	2000	5.505	5.073	00.74	6.117	6.117	6.390	6.520	6.526	6.368	7.103	7 .273	7.307	7.011	7	8.186	9.320	8.522	200.8	3.395		
1 #F5/295	SCALS SCALS SCALS SCALS SCALS SETERS/SEC		DIFFERENCE	-0.042	0.013	0.5.60	960.0-	-0.341	0.00	-0.058	-2.098	- C. 043	000	740.0	202.0	0.101	6.00	10.090	0.232	0.305	-0.075	-0.102	0.123	0.206	0.032	-2.082	-0.165	-0.047	50000	-0.342	0.514	10.555	0.200	-2.781	0.387	0.416 AMS
OIPOLE #257/11	00000000000000000000000000000000000000	SHOCK FRONT DATA COMPUTED FROM PARTICLE RELIES (197) + DESCRIPTION (197) + DESCRIPTION RELIES OF RADIUS SOURHED USING WILDHINGS OF RADIUS	MET 5KS	16.303	19.029	2000	24 . 3 < 5	24 . 125	25.793	27. 124	20.150	27.094	27.363	29.250	200.00	37.513	30.339	31 - 104	32.44C	32.440	33.102	33.079	33.379	35.413	35,483	36.474	36.590	37.537	33.165	39.596	40.053	40.738	41. :92	41.130	42.170	
	######################################	C+LCG(1+1)+ T=INVERSE G	FT CORS	16.045	18.017	20.844	24.422	24.667	2000	25.342	726.32	20.033	27.703	23.203	200.000	30.452	30.770	31. 455	32.233	32,135	33.470	33, 941	33.757	35.212	35.256	34.555	30.755	35.294	20.02	39.937	39, 539	41.292	40.04	42.765	42.389	
SHOCK FRONT DATA	A WAIENT TEMPERATURE TE #19.11 DEGREES CELSIUS AVAIENT PASSSONE PE 94.9 ALL UPASSALS ALL CHANGE TO CONTROL OF CANAGE ARABITATE TO CONTROL OF CANAGE ARABITATES	SFILLS RADVI	7-048 85.25	17.279	20.831	31.025	33.6 95	33.00	36.45	37.251	37.243	39.912	41.684	956.44	40.000	43.103	44.7.0	50.219	52.565	52.055	54.903	56.073	50.073	54.875	61.041	62.43	92.183	65.408	20000	70.340	71.458	73.634	74.590	70.410	73.423	

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631D 1220		MAN 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	444	444	m		A 1100 OF 1	1.004	1.915	1.950	1.949	1.375	2.033	2.055	2.080	2.072	2.007	2.031	
SMJKE PUFF		20 N 90					Sind of the second of the seco	, ,				-0	ω 4	٦.			٦.		v. ·
30.	2	72.00 1.00 1.00 1.00 1.00 1.00 1.00 1.00	44.0	200	1.35	13	ME ST 150 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	6.0	19.	552	500	49.	.71 .75	2 4	7.	1.710	1.627	1.338	EB VALUES S SCAL ING.
WF5/295	M 000.1 =	VELCEITY 1.324 1.381	1.375	0.871	1.117	= 2.000 M	VALUCITY VALUCITY 0.542 0.384	0.583	0.627	0.632	0.723	0.840	0.335	0.313	0.587	744.0	0.550	0.880	HWES SEAL
E 4351/11	CALED TINES	F0 0/4	J U 4	70-	0.54	CALED TIME	1000 1000 1000 1000 1000 1000 1000 100	00		00				00		0			8.0730 8.0733 ARE INVAR
01000	AT 5	MACH NOT 1.22 1.31	1.28	72.	60.00	ES AT S	1000 1000 1000 1000 1000 1000 1000 100	0.05	000	20.00	0.00	100	00.00	3.41	000	3.45	45.0	99.0	VALCED S
FIELD	S VILOCITIES	3CAL 1.050	1.057	1.002		E VILOCITI	74.15.00 1.00 1.00 1.00 1.00 1.00 1.00 1.0	1.001	1.524	6.813	0.335	2.050	1.914	1.655	1.285	619.0	0000	0.179	VALUES
VELOCITY	SANTICL:	35 35 AL 35 35 AL 10 2 2 2 2	1.203	4000	1.254	SANTICLE	10000	• •		1.515		1.737	1.711	1.735		1.714	1.000		AND CHANG

VELOCITY	r FIELD	OIPOLE	LE WEST/11	1 WF5/295	30.	SMUKE PUFF	GRID 1220			/A730123			
PART ICL	E VELOCI	TIES AT S	CALED TIME	= 3.000	S								
0.0.10.00	15.02.43.		MACH NO 000 000 000 000 000 000 000 000 000	7-1066	ME-SCAL 1.5743 1.549	なり につ こ こ こ こ こ こ こ こ こ こ い り し い り り り り り り り り り り り り り り り り	'n- • • •	0.13400		020000	24 24 25 25 25 25 25 25 25 25 25 25 25 25 25	Olumban	まり 命り のり 公司
01111000	4//02/200		00000000000000000000000000000000000000	74 0 7 4 5 5 7 4 7 5 7 7 7 7 7 7 7 7 7 7 7 7	1.797 1.748 1.748 1.757 1.764 1.963			004NO-0-		000000	00000000000000000000000000000000000000	My manua	m.n.n.a
224025-	400000			4000000	2.028 1.928 1.938 2.938 2.018	പപപപപ്ത		NIL VO DO 4		00000	0000000 0000000 0000000 00000000	4444444	4
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OBSERVE AND OBS	SERVED TIM	E VALUES VALUE AS SHOWN	= 8.0730 = 8.5933 ARE INVA	TIMES SCA TIMES SCA	LED VALUE R SCALING	ທ • • ນາງ ບ							

TABLE 7.2

VELUCITY	Y FIELD	DIPOLE	11/15= # 3-	*F5/295	30.	SYOK - DUFF	GRID 1220			/A73012	0		
PART ICL	E VELOCIT	TIES AT SC	ALED TIM	E= 4.000 M	S								
METERS	Y-504L	WACH NO	MACH NOT	PARTICLE VELUCITY	AF. TERS	7.7.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2	X-SCA ALTIRS	METERS METERS	MACH NOT	MACH NOT	PARJEYLE PARJEYLE	3-30AL	25000
1.089	0	17	0	0.335	30	2	2.590	1.021	0.0	0.0	0.0	5	
25.55	3.7	0.30	$\alpha \in$	0.400	00	V 4	2.591	0.829	04	000		U	
1.715	7	1	-	104.0) 1	1 4	2.530	004	4	0		3 5	
1.785	S	2	N	0.357	. 0	-	2.533	0.234	4	0.0	7		۰,
1.931	3	-	-	0.236	2	-	2.343	2.220	0	0	0	30	'n
1.420	0	c	0	6 40 0	D	-	2.459	1.992	10	0.0	·		ņ
2.053	0	2	-	0.241	0	4	2.363	1.302	w.	0	0		4
1.979		2	-	405.0		4	2.847	1.565	3	00	n	3	4
2.054	4	Q.	-	0.270	-1.		2.005	1.376	4	0.0			4
7000	N	1.	-	167.0	(2.761	1.218	4 :		*	٠,	
2.0.0	٠,٠	- 0	> -	101.0)		20132	0 40	•	•			٠.
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2.455		7	0	0.329	4) 4	2.725	0.117		0			יח ני
2.220	0	2	-	0.234	U.	4	2,988	2.184	æ	-	1)	0	2
2.262	ניו	3	0	0 • 29 4	7	1	2.963	1.958		0		Cu	S
2.234	7		0	0.363		1	2.975	1.750	30	0.1			*
2.327	3	7.	0	0.435	~	-	2.952	1.594	1.	0:0		3	7
2.320		4	0	C • 477	17.1		2.940	1.393		-			4
2.311	0	•	0.0	0.455	.)	-4	2.945	1.233	9			0	•
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2.462	17	7	0	0.232	S	-	3.110	2.152	0	0.1		-	0.00
2.401	7	4	0.0	0.412	4	-	3.093	1.946	J.	0		3	.s
2.459	9	.5	0.0	0.507	4	1	3.117	1.760	æ	0.0		-:	4
2.440		U.	0.0	0.563	4.		3.083	1.534	01	0		-	*
V	0	Ū.	•	710.0	4	-	2000	1.395	•	•		:	4
2.333	4 (•	0.363	4 .		3.063	1.211	0	-		-	4.
11.00		•	•	0.00	1	- ;	\$0.0°	1000	•		•	•	-
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2000		1		0.44.0			200	0			. `	•	, ,
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AND OUS	ERVED T	IME VALUE	A 25 1 NV AD	NI NI	ED VALUE	•							
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TABLE 7.3

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	•	200	2.32)	12	1.57	, E. C.	1	1,00		3
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			.13	n	~	• 50	4	•	•	4	-,
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OBSERVED DISTANCE VALUES 8.0730 TIMES SCALED VALUES AND OBSERVED TIME VALUE = 8.5933 TIMES SCALED VALUE. VELOCITY VALUES AS SHOWN ARE INVARIANT UNDER SCALING.

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/A760120		WERY/RE	OC	0	0.0	0.0	000	0	0.0	0.0	00	•	0	0.0	0.0	0.0	0.0	0	000	, ,	00	0.0	0.0	0.0	000	000	00	0.0	0	000	00		0.0	0.0	0.0	0	000) -	0.2	~
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JFF GRID 1220		4X	11.4	4	2	5	47	0	3	40	200	7	3	55	57	Sa	20	8	00	2 0	10	72	72	7	0.0	0 0	32	9	00	30	0 0	84	33	53	32	20	200	200	0	0.0	3	5
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30.	SΣ	A-SCAL	2.945	2.074	2.910	2.199	2.763	2.790	2.745	3.077	3000	200	440.5	3.043	3.005	3.033	2.936	3.022	5000	2000	3.040	3.223	3.287	3.232	3.191	1 9	3.224	3.145	3.445	3.333	3.331	3.407	3.452	3.451	3 . 34 3	3.33	3.573	3.503	3.531	3.523	3.553	3.554
NF 5/295	000 • 9 =	VARTISTY	300	2	.22	0	51.	31.	.18	.17	010	, ,	.21	. 22	7 .	. 2	• 17	7	200	21	25	.24	.25	17.	. 23	24	30	. 29	. 24	. 23	282	67.	. 32	.33	.3.5		0 .		.25	. 27	67.	. 32
LE #EST/11	CALED TIME	WACH NU	40.0	0.05	0.00	0.03	10.01	-0.03	0.02	0.02	000	90.0	0.09	60.0	0.05	10.0	10.0	20.0	000		000	0.02	C. 03	0.03	000		-0.01	-0.00	0.05	0000	0.02	0.03	0.00	00.0-	-0.05	0.00		40.0	0.01	00.00	-00.00	-0.01
010	TIES AT S	MACH NO					::	7	-	-	ci.		7	.2	-			7	20		101		::	.1.		4				2.	ייי		m			•	? .	١٣.	2	2		.,
FIELD	VELOCI	Y-SCAL	22	.75	.03		20	.42	57.	42.	000		.39	. 42	2.2	33	0	1	200	3	.73	.58	.37	. 42	2 .		4	.27	5	100	56	.30	• 10	00.	. 8.	0 4		17	76.	.75	.56	• 36
VELOCITY	PANT ICLE	X-SCAL METERS	2.915	2.073	2.333	06/10	2.320	2.755	2.734	3.051	0.00	3.077	3.000	2.382	8000	·i ·	2.770	•	3.7.0	3.2.3	3.235	3.211	3.252	3.221	3.179	3.150	3.191	3,133	3.429	3.337	3.378	3.572	3.383	3.340	3.342	3.330	3.414	3.548	3.530	3.5.33	3.542	3.521

OBSERVED DISTANCE VALUES 8.0730 TIMES SCALED VALUES AND CBSERVED TIME VALUE = 6.59434 TIMES SCALED VALUE. VALUES AS SHOWN ARE INVARIANT UNDER SCALING.

TABLE 7.5

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11.5.79	11.752 1.5	87 1.97	-	0		2	n n	200	20		0	• •	. 0	* 4
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22.000 0.195 3.519 5 5 4 1133 1.335 0.35 0.393 0.393 1.375 0.204 0.000 0.204 3.519 5 5 4 1145 0.204 0.205 0.	2.27 0.00 0.00 0.00 0.00 0.00 0.00 0.00	12 0.26	-		:-	22	'n		20		0	• •	:-	*
1.777 0.22 0.02 0.224 3.476 4 4.1150 0.1950 0.49 0.49 0.49 0.224 1.575 0.224 3.476 4 4.1150 0.239 0.44 0.000 0.224 3.476 4 4.1150 0.224 0.039 0.44 0.000 0.224 1.575 0.224 3.5476 4 4.1150 0.239 0.44 0.000 0.324 0.000 0.000 0.324 0.0000 0.324 0.000 0.324 0.000 0.324 0.000 0.324 0.000 0.324 0.000 0.324 0.000	1975 1975	2.20	-	•	-	10.	S	13	33	•	0	•	-	7
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58 2-160 0.21 0.03 0.211 3.771 5 4.393 1.185 0.45 -0.04 0.450 4.4 86 1.972 0.22 0.03 0.22 0.23 3.744 4 4.353 0.992 0.33 -0.03 0.392 4.4 43 1.764 0.24 0.05 0.263 3.757 4 4.350 0.643 0.43 -0.07 0.432 4.4 45 1.552 0.25 0.05 0.263 3.757 4 4.350 0.643 0.43 -0.07 0.432 4.4	58 2.160 0.21 0.03 0.211 3.771 5 4.393 1.185 0.45 -0.04 0.463 4.443 4.372 0.03 0.22 0.22 0.22 0.23 3.744 4 4.373 0.992 0.55 -0.03 0.392 4.457 4.357 0.02 0.25 0.20 0.20 3.757 4 4.355 0.643 0.03 0.39 -0.07 0.432 4.472 4.352 0.25 0.20 0.20 0.20 3.757 4 4.355 0.643 0.043 -0.07 0.432 4.472 4.345 0.443 0.055 -0.09 0.559 4.368 4.368 4.345 0.443 0.055 -0.09 0.559 4.368 4.368 6.357 VALUES # 6.5933 TIMES SCALED VALUES # 6.5933 TIMES SCALED VALUES # 6.500 0.550 0	94 6.27		0		00	, w	39	30			•		4
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45 1.552 0.25 0.05 0.263 3.757 4 4.350 0.643 0.43 -0.07 0.432 4.4 32 1.360 0.26 0.03 0.260 3.754 4 4.346 0.431 0.55 -0.09 0.559 4.3	45 1.552 0.26 0.05 0.263 3.757 4 4.365 0.643 0.43 -0.07 0.432 4.412 4.365 0.643 0.655 -0.09 0.559 4.368 4.368 0.851 0.55 -0.09 0.559 4.368 0.852 0.853 0.853 0.853 0.853 0.853 0.853 0.853 0.855 0.853	43 1.76	30	•	20		Ω 4	25	2.4		000	•		*
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	RVED DISTANCE VALUES = 8.0730 TIMES SCALED VALUE OBSERVED TIME VALUE = 8.5933 TIMES SCALED VALUE OF VALUES AS SHOWN ARE INVARIANT UNDER SCALINS	32 1.30		•		. 70	4	• 34	4 3	•	0.0	•	•	m
		CITY VALUE	AS SHUR	ARE IN	AIANT UND	SCALI								

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TABLE 7.6

THIS PAGE IS BEST QUALITY PRACTICABLE FROM COPY FURNISHED TO DDC

VLL 301 T	Y FILLD	0.450	LE *25T/1	11 AF5/295	.05	SHUK PUF	F Gr 10 122	0					
JANT ICL	= VELDET	TIES AT S	CAL TO TIM	090.6 =5						74730120	0		
i	Y-5.4				2								
3.1.5	Z:1.3	TACA S	MACH		X-SCAL	2000	SCAL		1				
25	340.5	200	00.00	133	3.195)	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	SKIT S	2.00	つていいてい	-1>	3-SCAL	SE SE
200	240	***	-0.01	137	4.203	U 4	121		0.50	00.00	1.90	401.4	10
00	604.1	2000	***	131	3.204	• •	000		0.20	0.0	203	2000	0
20	1.235	0	0.0	000	3.0.47	4	133		2.51	50.0	211	4.127	d
	1.047	63	40.0	187	2000	4.	120		17.0	50.00	215	4 - 157	1 4
14	2000		90.0	157	3.130	•	120		0.24	00.0	20.00	4.178	4
	0.034			164	3.051		000		0.23	00.0	277	4.504	4 1
7	6.288	000	500	000	3.046	-	260		0.22	-0.01	216	4.145	7
2;	2.550	0.12	0.03	200	3000	·7 i	100		70.00	-3.02	273	4.115	n ~
* *	2.001	0.11	00.0	125	345	n ı	25.3		0.50	200	207	4.076	•
7		30.00	0.05	050	3.364	0 4	233		0.21	0.05	200	4.209	:0
4	1.596	000	400	113	3.350	• •	600		0.22	0.03	220	4.539	'n.
000	1.249	0.0	000	0 1	3.335	4	740		0.24	0.03	245	4 . 201	4 .
00	1.043	60.0	200	200	3.302	4	100		0.65	0.01	247	4.271	* 4
0	6.050	20.0	0.00	000	2.000	4	263		000	10.0	797	4.324	1 4
*	50000	1:0	0.10	200	2000		240		0.00	00 -0-	507	4.343	,
5	010	20.0	41.0	200	3.441	-,-	213		20.00	10.01	255	4 . 320	. 6
	20.00	00.00	80.0	105		n ~	550		00.00		200	4.255	3
000	1000	*	90.0	150	3.577	2	308		0.21		573	4 - 238	٣
	000	05	0.05	153	3.523	n c	101		9.22		612	4.377	:0
27	010	51.0	6.05	153	3.507	0 4	371		0.25	3.02	177	70 4 . 4	n
22		***	10.0	143	3.537	1 4	189		0.26	0.0	0 0	4 - 375	4
2.0	000		0.05	135	3.537	+ 4	000		2.27	0.00	200	4.399	4
5.5	17	000		155	3.597	. 7	273		67.0	0.02	100	4.397	4
5.5	0.0		200	940	3.562	4	200		0.27	0.01	273	2000	4
5	0.652	200	2000	101	3.453	1	200		0.27	-0.02	27.2	2000	d :
2	0.400	0.11	5.0	000	3.235	27	37.0		0.24	0.01	543	4 - 423	+ 1"
0	267.0	-0.06		010	3.021	m	332		0.00	-0.03	304	4 . 39 3	? "
7.0	2.205	0.14	0.00	200	2010	m	205		200	0.03	331	4.341	m
0 4	5260	0.17	90.0	83	3.077	n ı	111		0.23		553	4.514	0.0
200	000		50.00	11	3.636) 4	020		0.28		222	4.511	S
15	300	000	40.0	69	3.711	, 4	112		0.23	0.00	0 0	4.551	4
20	1.150	0.00	20.00	21	3.715	4	200		0.30	0.0	200	4.023	4
2	1.011	0.03		40	3.759	4	200		3.34	ċ	340	4.578	t <
0	0.816	0.11	0.03	00	3.700	4	503		0.33	-2.00	177	4.594	, 4
	0.045	0.11	6.05	* 1	401.0	41	20				73	4.020	1 4
٥.	0.452	0.13	0.07	4.5	3000	,	- 35		3.00	•	137	4.552	m
٥.	0.280	0.18	90.0	88	3.687	· ·	00		0.40	:	55	4.511	*
. 1	20173	9:00	90.0	70	3.334	, ,	12		0.30	00.0	200	4.514	m
1	1.732	0.00	0.05	53	3.850	, .	500		0.23		10	1800	n
6	1.572	2000	000	8	3.813	4	23		0.36	0	2	5003	n ·
0:1	1.309	0.15	0.00	0 1	3.030	4	53		0.32		10	4.004	1 4
0	1.172	0.16		22	3.4.00	*	55		35.50		28	4.679	4
o u	0.00	E	00.0-	30	3.925	+ 4	co		0.33		25	4.700	4
	2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	0	10.0	40	3.983	**	200		0.34		200	4.712	4
200	0.455	21.0	000	503	3.437	3	000		0.40	-0.03	0 0	4.743	4.
0	0.280	0.21	2000		3.011	2	10		0.45		52	4.640	2
0	2.179	0.18	200	77	3.805	'n	90		0.30	•	69	4.603	7 ~
211	1.964	61.0	0.05	0 ~	3.043	s .	63		000	0.00	29	4 . 792	0 (
	1.756	61.0	0.05	25	3.000	ກ ເ	69		0.35		50	4.766	0.0
•	0000	1100	0.04	10	3.040	, ,	7.5		0.35		2 1	162.0	4
	215		0.02	2	3.96.5	14	6.4		6.35		10	4.786	4
	2000	0.00	3	68	4.024	4	2		0 4 3		00	4.775	4
	60000		000	70	4.025	4	+ 4		0.45		22	070.4	•
	0.026	41.0	•	25	4.075	4	0.10		0.47	0.0	200	4.878	•
	6.449	0.21	0.0	0 -	3.992	n :	24		0.00		25	4. 794	+ m
0					204.0	m			1	•	34	4.757	m
200	LISTANCE	VALUES=	8.0730 T	IMES SCALE	251.120								

TABLE 7.7

DESTRUD LISTANCE VALUES= 8.0730 TIMES SCALED VALUES AND CREENCTY VALUE = 8.5933 TIMES SCALED VALUE. VELOCITY VALUES AS SHOWN ARE INVARIANT UNDER SCALING.

	ነር) mD ሌህ ፌክላ ፋፋሾህፋ ፋፋላር ኮ ላ ኔቀፋ ሲ ሶ ላቀፋ ፋ ህ ኮ ጳቀፋ ቀ ህ ኮ ጳቀፋ ቀ ላር ኮ ላ ፋቀቀ
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	44 E10000303000030000000000000000000000000
/A730120	\tag{\}\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\
	AEGOOGOOGOOGOOGOOGOOGOOGOOGOOGOOGOOGOOGOO
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CKID 1220	$\frac{\chi_{X}}{167444444444444444444444444444444444444$
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	######################################
. S. W.	######################################
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CALED TIME	>E
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VELUCI	\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\
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TABLE 7.9

		25.03 25.03	ດທ	đ	4	*	1	200	4	4	4	4	:0	0	4	4	*	4	2	s	4	4	4	4	n	ın	4	4	4	4		
		3-SCAL	4.000	4.890	4. 354	4 . 390	2.010	4.330	5.015	50005	5.004	5.077	5.176	5.162	5.181	5.194	5.198	5.233	5.330	5.305	5.331	5.330	5.354	5. 382	5.436	5.450	5.470	5.446	5.496	5.510		
		PAESSES PE	:-	-	•	-	: 7	::	-	-	-	-	-	-	-	-	-	7	-	-	-	-	-	:	-	-	-	-	-	7		
14740120		WACH NO	00		•	•		0		0	0	0	0	•	0	•	0	0	0	0	0	0	0	0	0	•	0	0	0	0		
		MACH NO	000	-	-	:	: -	-	-	-	-:	-	-:	-	-	-:	-	-	-	-	-	-:	-	-	-	7	-	-	-	-		
		ME 1904	1.944	1.743	1.0.44	1.370	2.148	1 . 958	1.738	1.535	1.360	1.133	2.125	1.938	1.751	1.571	1.356	1.182	2.107	1.904	1.734	1.540	1.363	1.156	2.144	1.910	1.729	1.561	1 • 365	1.150		
GRID 1220		M TOCA	4.359	4.38	# (D) • # ·	4 - 365	200.5	4. 379	5.014	4.995	4.980	5.034	5.109	5.101	5.180	5.187	5.175	5.190	5.324	5.305	5,333	5.321	5.332	5.337	5.423	5.450	5.469	5.473	5.474	5.465		
SMOKE PUFF		200	ດເລ	4	4.	4 4		2	4	4	4	4	2	2	4	t	4	4	S	S	4	4	4.	4	2	S	4	4	4	4	ø•	
30.	SW	A-SCAL F	4.070	4.062	06000	4.055	4.231	4.251	4.209	4.242	4 .275	4.297	4.417	4.370	4.393	4 . 402	4.403	4.454	4.541	4.561	4.536	4.560	4.550	0 × 0 · +	4.639	4.084	4.732	4.698	4.723	4.766	VALUE VALUE	n
MF5/295	= 11.000	VELUCITY	0000	0.072	0.123	0.120	90.0	0.039	0.051	6.063	260.0	0.118	0.080	0.037	0.054	0.061	9.095	0.117	0.072	0.064	290.0	0.072	860.0	0.110	0.095	0.043	0.081	680.0	0.100	0.118	TI MES SCALE	N N
E WEST/11	CALED TIME	LO.	00	0		0	, 0	0	0	°	0	•	0	0	•	•	•	0	0	•	0	•	•	•	0	0	•	0	•	•	8.5933	N .
DIPOLE	S AT S	AACH NOT	20	90.0	0.0		0.03	40.0	0.04	90.0	0.03	01.0	10.0	0.0	0.05	20.0	60.0	01.0	0.07	0.00	0.07	20.0	60.0	0.10	60.0	0.03	0.03	60.0	60.0	0.11	VALUE VALUE	AS
FIELD	E VELOCITIE	Y-SCAL MITCHS	1.996	1.776	1.044	1.395	2.175	1.978	1.770	1.586	1.396	1.218	2.176	1.953	1.748	1.334	1.359	1.184	2.173	1.974	1.764	1.576	1.373	1.197	5 + 1 + 2	46.	.76	55	.37	61.	25	VALUE
VELOCITY	PARTICLE	X-SCAL METCRS	4.077	4.062	150.4	040.4	4.618	4.249	4.208	4 . 2 34	4.250	4 . 250	4.106	4. 375	4.378	4.391	4.376	,	•	095.7	;	4.351	4.525	4.544	610.4	4.033	4 . 7 31	4.639	4.699	4.721	AND CHELL	וְ

TABLE 7.10

74740120	AL MARK TO THE TOTAL THE T	
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TABLE 8.2

TABLE	8.3													
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2.911	O	D. 0 44	2.911	-	7	•	67.		4	•	1.289		3.463	•
2.342	0	1.033	2.338	-	0	•	44.		4	•	1:1:1	4	3.370	•
2.868	0	1.021	2.831	-	0	•	.24		-	•	0.913	10	3. 304	7
2.360	O	1.043	2.838	3	0	•	.32		6	•	0.734	.5	3.852	
3.148	N	101-1	3.159	S	3.3	•	04.		٦		0.567		3.310	
3.153	-	1.222	3, 153	2	0	•	.20		'n	•	0.382		3.779	
3.147	-	1.377	3.152	4	00		.12		s	•	2.025	9	3.973	
3,140	-	1.184	3,103	4	00	•	.55		4	•	1.820		3.965	7
3.114	-	1.010	3,162	4	63	•	.58		4	•	1.639	•	3.907	•
3.096	7	1.129	3.180	4	21	•	. 50		4	•	1.451	0	3.374	4
3.083	O	1.348	3.089	_	1	•	.36		4	•	1.205	.0	3.909	•
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m	-	1.305	3,359	4	2	•	.13		4	•	1.635	-	660.4	
'n	-	1,233	3.303	4	5	•	• 56		4	•	1.445	-	4.105	•
3.250	0	1.077	3.250	-	3	•	99.		4	•	1.271	•	4.126	
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TABLE 8.5

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TABLE 8.6

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TABLE 8.7

JUSERVED DISTANCE VALUES = 0.0730 IIMES SCALED VALUES AND COSCAVED TIME VALUE = 0.5933 TIMES SCALED VALUE. DENSITY VALUES AS SHOWN ARE INVARIANT UNDER SCALING.

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TABLE 9.2

JBSERVED DISTANCE VALUES 9.0730 IIMES SCALED VALUES AND DESERVED TIME VALUE = 8.5933 TIMES SCALED VALUE. PRESSURE VALUES AS SHOWN ARE INVARIANT UNDER SCALING.

Carlotte State of the state of

PRESSURE FIELD	DIPOLE	EST/111	DIPOLE WEST/11 WF5/295 30*	30.	SMUKE PUFF GRID 1220	/A730120
AVERAGE HYDROSTATIC OVERPRESSURES AT SCALED TIMES 6.000 MS	OVERPRE	SEURES AT	SCALED	TIME	6.000 45	

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MACH 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4	3.502	3.515	3.532	3.411	3.454	3.418	3.514	3.602	3.597	3.604	3.027	3.657	3.677	3.627	3.574	3.544	3.740	3. 731	3.721	3.722	3.738	3.772	3.807	3.743	3,691	3.656	3.853
520000 50000 50000 50000	0.148	0.466	0.693	0.364	964.0	0.635	0.322	0 . 20 3	678.0	0.938	162.0	0.564	0.347	0.565	266.0	0.938	0.702	0.343	0.175	0.215	0.401	1.074	1.499	0.940	0.819	1.097	0.721
MF TEAS 1.653	-	-	0	0	0	0	C	_	-	~	-	~	C	0	0	0	CJ	-	-	-	-	-	0	O	C	0	C
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METERS 2.983	2.988	2.911	2.874	2.591	2.838	3.159	3.153	3.152	3, 163	3.152	3.180	3.089	3.079	3.092	3.005	3.328	3, 316	3,323	3.324	3.359	3, 363	3.250	3.246	3.306	3.254	3.430	3.467
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X AND Y LOCATE THE CENTER OF A PLANE QUADRILATERAL WHICH IS A CELL OF 4 NAIGHBOURING SMOKE PUFFS. DVZRPRESSURE IS AVERAGED OVER THE AREA OF THE CELL AND IS EXPRESSED AS A RATIO TO THE AMBIENT PRESSURE. DBSERVED DISTANCE VALUES = 8.0730 TIMES SCALED VALUES AND OBSERVED TIME VALUE = 8.5933 TIMES SCALED VALUE PRESSURE VALUES AS SHOWN ARE INVARIANT UNDER SCALING.

PACSSUR	E FIELD	OIPOL	E 455T/11	11 WF5/295	30.	SAUKE	PUFF GRID	1220		/ 47	A730120			
AV 33 A GE	HYDROSTATI	FATIC SVEAP	RESSURE	S AT SCALED	TIME =	7.000 MS	•							
X-3CAL	Y-SCAL	PRESSURE	A-SCAL METERS	200	X- SCAL	ME TOAL	PAFSSUFC	100 A L	500 500 500 500	X X X X X X X X X X X X X X X X X X X	X-1SOAL	PRESTA	100 M	ASSESSE POSSES POS
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TABLE 9.4

OBSERVED DISTANCE VALUES = 0.0730 TIMES SCALED VALUES AND CASERVED TIME VALUE = 8.5933 TIMES SCALED VALUE. PRESSURE VALUES AS SHOWN ARE INVARIANT UNDER SCALING.

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93 0.096 3.653	4	1.08	69.	•	4	•	60.		75
23 0.203 3.656	7	0.89	.28		4	•	06.	•	.79
39 0.099 3.619	4	0.71	.31	•	3	•	. 72		17.
54 0.265 3.601	4	0.51	.23	•	2	•	.54	•	17.
56 0.076 3.583	7	2.02	.35	•	ഗ	•	.34	•	.68
81 0.159 3.763	4	1.83	.23	•	+	•	.01		34
39 0.073 3.757	4	1.63	.42	•	4	•	.82	•	.34
87 0.577 3.760	4	1.44	.35	•	4	•	.63		. 85
31 0.417 3.770	7	1.26	.34	•	4	•	44.	•	. 35
55 0.348 3.797	4	1.07	.65		4	•	.27	•	. 86
57 0.020 3.822	7	0.39	.35	•	4	•	600	•	68.
01 -0.159 3.836	7	0.10	77.	•	6	•	06.	•	. 32
31 -0.063 3.790	7	0.51	.25	4.324	٣	•	.72	•	88

X AND Y LOCATE THE CENTER OF A PLANE QUADRILATERAL WHICH IS A CELL OF 4 NEIGHBOURING SMOKE PUFFS. OVERPESSORE IS AVERAGED OVER THE AREA OF THE CELL AND IS EXPRESSED AS A RATIO TO THE AMPIENT PRESSORE.

DBSERVED DISTANCE VALUES - 0.0739 TIMES SCALED VALUES AND COSERVED TIME VALUES = 0.5933 TIMES SCALED VALUE. PRESSURE VALUES AS SHOWN ARE INVARIANT UNDER SCALING.

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98286	SURE FICED	OIPOL	OLE #851/11	11 #F5/295	30.	SHUKE 3	UFF GRID	1220		147	90120			
AVER	AGE HYDRUST	ATIC OVER	PRESSURE	S AT SCALED	TIME	9.000 AS								
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X AND Y LOCATE THE CENTER OF A PLANE QUADRILATERAL WHICH IS A CELL OF 4 NEIGHBOURING SMOKE PUFFS. OVERPRESSURE. OBSERVED DISTANCE VALUES= 8.0730 TIMES SCALED VALUES AND OBSERVED TIME VALUE = 8.5933 TIMES SCALED VALUES PRESSUME VALUES AS SHOWN ARE INVARIANT UNDER SCALING.

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OBSERVED DISTANCE VALLES = 8.5033 FIMES SCALED VALUES PRESSURE VALUES AS SHOWN ARE INVARIANT UNDER SCALING.

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MARTIC JUENCREES AT SCALED TINTE 12.000 MS	### SSURES AT SCALED TINGE 12.000 MS ### SSURE ATTC JVENPRESSURE AT SCALED TINGE 12.000 MS ### SSURE ATTS				667/6-W 11	2			022:		/47	74750120			
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